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AVIATION

November 1934

THE LARGEST AMERICAN AERONAUTICAL MAGAZINE

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OF LEADS OF AIRCRAFT

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BALTIMORE, MARYLAND, U. S. A.

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Up from Chaos

ONCE again we sit down to record air transport history. Perforce it has been our practice to devote one issue of AVIATION to review the accomplishments of American airlines, to present to the reader an up-to-date picture of their statements. Heretofore the record has been one of continuous progress. All statistical methods have stood up to spectacular rates. The past year, however, has witnessed a series of events that were as fantastic as they were unpredictable.

We have had to make drastic revisions in many of our older concepts, and some fundamental changes of our ideas as to the future. Scarcely are turning points in history as clearly defined as was that marked by the cancellation of the air mail contracts in February last. An entire industry is rapidly retooling; our nation is filled with an insurmountable debt. Everything seems to be at a full stop. The crash left those who had spent years building up a complicated and expensive machine sitting in the midst of a tangled heap of wreckage, their hands still clutching bits of control wheels and levers. For better or for worse, they had to land themselves pitifully out of the ruins and begin to lay out the new work of salvage and reconstruction.

To have attempted to present any picture of American Air Transport at any earlier period of this year would have been a hopeless task. Now that the dust has begun to settle, however, the outlines are becoming a little clearer, the new pattern is beginning to emerge. The airplane (page 246) has taken what appears to be a permanent set, and, although shifts in ownership and control of certain lines may be expected in the near future, the general operating picture has attained some degree of stability. Most of the old lines, in commercial and private forms, have resumed business on their old stands, and the newcomers have gotten down to the crucial postnatal days, and are beginning to formulate ambitious plans for the future. Everywhere in this year will be found a

tableted record of what's who on the lines, a listing of the equipment in current use, and data on the mileage flown and territory covered. To have a degree of perspective to our readers, a grasp of visitors from overseas have given us their impressions of air transport in America.

Although individuals have suffered heavily by the setbacks of 1954, and many an airline, new and old, is struggling along in the red under the double burden of continued losses and heavy development expense, we cannot help but feel that the future may be faced with some confidence. 1) For events of the past year had no other result but the focusing of public and government attention on a long-range planning program for the air transport industry, looking toward a broad policy of coordination with other elements in the nation's transportation scheme, it would have been well worth the end. The industry is moving from the ashes of the spring of 1954 and stands poised for take-off in a flight that will set new records in the transportation history of the world.



Spares in military airplanes replaced the single-aisle and dual previously used, mail, passengers, and cargo have arrived on all of the routes named type scheduled daily. In the spring of 1953, Ford in-military transport replaced the biplane, and at the end of that year five cargo types were being flown daily.

Following the acquisition of all air mail, air and contract, operations were somewhat curtailed between February and May, 1954. During the latter month, however, passenger and cargo services were extended to Detroit, Mich., in order that operations could continue to the Detroit area. The Post Office department between Detroit and Washington, via Pittsburgh, Albany and Cleveland.

Although the Pennsylvania Airlines and Transport Company did not receive the award of the contract for air mail service on the route which its predecessor company had developed for more than seven years, service was continued to four round trips daily, with passengers and cargo. The company, however, was awarded an air mail contract on AMJL between Detroit and Cleveland.

One of the features of the operation in that between Cleveland and Detroit is a one-way flight over Lake Erie and into Detroit. This route was adopted only after several tests with airplane loads which showed that a high altitude could be satisfactorily maintained in event of failure of one engine, and that land could be reached from any point on the route in event of failure of two engines.

After several months of this type of operation it was decided to replace the Lockheed Constellation, being used between Detroit and Milwaukee (including a 10-day flight over Lake Michigan), with Ford trimotors. These planes are now being equipped with Rotolux gear to be an auxiliary landing, and will be in service on May 1.



Sub Lake-Genet Falls Airline

Operations of the Sub Lake-Genet Falls airline were originally scheduled to begin in November 1953. The airline, and scheduled, was scheduled to begin the operation of the air mail contract in February of this year. An effort to continue the operation of the route on a reduced schedule as a gesture and to provide service only was made for a short period, but it proved to be a losing venture, and operations were suspended on March 30 of this year.

In response to a development of the route, a bid was submitted by Mr. Frank previously, and contract subsequently awarded to him, resulting in the resumption of operations on May 15, 1954. In August, the contract was extended, under which operations are being performed at this time.

The Sub Lake-Genet Falls airline, by the construction at Erie, Pa., and Sub Lake City with two of the scheduled biplane round-trip routes. The scheduled office at Montreal, Quebec and Utah a connecting link to Utah cities through the United States. The route is a direct over the Atlantic Ocean service, ranging



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(President)



C. B. Cohen
(Vice President)



R. G. Kennedy
(Vice President)



John W. Wilson
(Vice President)



C. B. Cohen
(Vice President)



John W. Wilson
(Vice President)



John W. Wilson
(Vice President)



John W. Wilson
(Vice President)



John W. Wilson
(Vice President)

in addition from 10,000 to 12,000 ft above sea level, which requires the employment of highly skilled personnel and the use of the most suitable equipment. Negotiations are now in process for the acquisition of multi-engine planes by the end of the year, or, earlier, if possible.



TWA, Inc.

ONE of the outstanding developments of this year has been the establishment of overnight transport schedules between the Atlantic and Pacific Coasts, made possible by the acquisition of TWA Trimotors, developed by the standard engineering resources of the TWA technical staff and the Douglas Aircraft Corporation. This transport has supported all passenger loads for speed and economy (the 228 mph, cruising speed at 12,000 ft, enables TWA, to maintain the fastest air transport schedule in the world).

The Douglas biplane New York-Los Angeles at 4 o'clock and arrival at Los Angeles at 7 o'clock the next morning with only three intermediate stops. Less than two weeks after this schedule was inaugurated, the demand for service became so great that a second schedule, the Comet, was inaugurated, leaving New York at 8:30 in the evening, and arriving in Los Angeles the next morning at 9:15. In reverse direction, the main-line planes leave Los Angeles at 4 p.m. and 9 p.m. and make the journey to the Atlantic undisturbed in seven hours.

Powered with two 1700-hp, supercharged Wright Cyclone engines, and fitted with controllable pitch propellers, these two new biplane units at altitudes between 10,000 and 12,000 ft.

The new biplane Comet, specially designed and fitted, which is now being flown from Los Angeles to New York. In direct hours and 30 minutes by just one intermediate stop at Salt Lake City. It is being equipped with all the latest developments for high altitude flying and is being used at the Kansas City base as a biplane, having already made several trips to the contract in other routes.

With one or two stops, the biplane at altitudes ranging from 15,000 to 20,000 ft or more. The Douglas trimotors, the entire system, the old Ford biplane has completely replaced by the new biplane early in September of this year. Separate mail and express schedules at night between New York and Kansas City are maintained with Northwest Dakota and Mustangs.



United Air Lines

UNITED AIR LINES, Inc., represents a combination of four companies, the Eastern Air Transport, Pacific Air Transport, National Air Transport and the Valley Air Lines. It also operates in 30 cities, situated in 10 western states, and one foreign country. It is the largest operator in the United States from the point of airplane home, having recently flown its 70th million miles, and its 200,000,000 miles were at night.

Used to be now flying at the rate of approximately 19,000 hours a year, of which 15 per cent is strictly a passenger service. Most is moved on per person basis of its daily schedule. United is affiliated with the Western Express Agency in the management of its fleet.

United is now operating without mail contracts, a passenger agency since 1953, with United Lines in the Eastern States, Chicago and Kansas City, Spokane, Seattle, Cleveland and Philadelphia, and all the rest of its routes it has been operating from since to October, in which case, having been the last responsible for the mail and contracts, when bids were originally called for, and again when bids were called for, after the February cancellation of mail contracts.

The company is now operating Boeing 200's and Douglas C-47's, and is also operating (2458 hp) Supercharged Wasp motor mail airplanes on all routes. It is offering basic service and is disposing of

its former fleet of 75-engine and single-engine planes, except a reserve of single-engine planes, to carry mail when service is required, and per person operation of passenger planes.

United operations set-ups calls for United Lines at Chicago in the Eastern States, Chicago and Kansas City, Spokane, Seattle, Cleveland and Philadelphia, and all the rest of its routes it has been operating from since to October, in which case, having been the last responsible for the mail and contracts, when bids were originally called for, and again when bids were called for, after the February cancellation of mail contracts.



Flying Air Service

WYOMING AIR SERVICE was incorporated the early part of 1954. Regular flights will be made between Laramie, Wyo., and Denver, Colo. The company was formed after two Wyoming

Stinson Juniors. Within three months, three stops and two pilots were added and the line was expanded to run north and south.

After three years of operation, the company still remains in the first class of the industry. The company is now operating with a fleet of 10 Stinson Juniors, with considerable pilot personnel and several high-powered engines, were purchased. The company is now operating with a fleet of 10 Stinson Juniors, with considerable pilot personnel and several high-powered engines, were purchased. The company is now operating with a fleet of 10 Stinson Juniors, with considerable pilot personnel and several high-powered engines, were purchased.

Mail-carrying equipment is being added to the fleet. The company is now operating with a fleet of 10 Stinson Juniors, with considerable pilot personnel and several high-powered engines, were purchased. The company is now operating with a fleet of 10 Stinson Juniors, with considerable pilot personnel and several high-powered engines, were purchased.

The Aeronautical Council of Commerce

An account of its varied activities in the air transport field

THE Aeronautical Council of Commerce is an advisory body to the Federal Aviation Commission, and is a body of representatives of the air transport industry in all branches of the industry, and to its own members in all branches.

Another committee of the Council is the Air Transport Committee, of which C. B. Smith, vice president of operations of American Air Lines, is chairman. This committee, which is composed of 30 representatives and their chief pilots, was held in Washington in August, 1954, and was the first meeting of the Council since its formation. The committee is now operating with a fleet of 10 Stinson Juniors, with considerable pilot personnel and several high-powered engines, were purchased.

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(Kearney), Rensselaer (American), Kennedy (United) and Taylor (General), given a good deal of the scope of the committee.

(a) Educational campaign for promotion of air passenger and express.

(b) Sales information for local business.

(c) Report on development of air mail.

(d) Uniform listing for air line information in telephone directories.

(e) Uniform listing for air line information in telephone directories.

(f) Traffic agreements, including free transportation to Air Transport Code.

(g) Traffic agreements, including free transportation to Air Transport Code.

(h) Traffic agreements, including free transportation to Air Transport Code.

(i) Traffic agreements, including free transportation to Air Transport Code.

(j) Traffic agreements, including free transportation to Air Transport Code.

(k) Traffic agreements, including free transportation to Air Transport Code.

(l) Traffic agreements, including free transportation to Air Transport Code.

(m) Traffic agreements, including free transportation to Air Transport Code.

(n) Traffic agreements, including free transportation to Air Transport Code.



Work now under way by the Bureau of Aeronautics to improve and to extend radio aids to air navigation is discussed

By Rex Martin

Assistant Director of Air Commerce is chief of Air Navigation

AN OCEAN LINER maintains contact with the shore and other ships through its radio equipment. As radio equipment is key carry as many as five. On an air liner, the duties of "operator" usually are performed by the skipper and his crew, that is to say, by the pilot and co-pilot.

Thus, in developing aeronautics radio facilities for the Federal Aviation System, the Bureau of Air Commerce has kept constantly in mind that these facilities are to be used by pilots, who save a multitude of vital duties to their home. Now are they installed solely for the use of the highly trained aviators who fly with passengers and mail. Many private flyers are now equipping their airplanes with radio so that they too can follow the courses marked by the radio range beacons, listen to the broadcasts of weather information, and communicate, if necessary, with ground personnel of the Air Commerce Bureau's Air Navigation Division. In such instances, we have to remember that: (1) we must make the operation of radio aids as simple as possible from the standpoint of reception in the cockpit of an airplane, and (2) improvements must give accuracy not of present procedure and position, clear tables, aerial changes

would confuse the users of aeronautical radio. Sometimes an improvement in the department's radio system is apparent to the pilot only as an increase in efficiency and dependability of a service with which he is entirely familiar. For example, radio range beacons continue to be located at night. Mutual coordination between transmitters of radio signals caused the courses to swing away from the towers over which they were intended to fly. This was solved by pointing the antennas with T-2, or modified Alouette type antennas which consist of four distinct steel towers, each connected with the transmitter by an underground cable.

High towers have no place on the edge of an airport. This was an objection to the old loop antennas, and is even more involved with the new type. Accordingly the bureau has adopted the process of establishing radio stations 2, 3 or 4 miles away from airports, where there can be no question of obstructing aircraft while landing and taking off. Such an arrangement, however, made it difficult to coordinate the work of operators, officers, pilots and bureau personnel. There was a constant pouring of messages between airport and radio sta-

tion. In emergencies, delays of even a minute or two may be serious. Also, many operators were required, for in most cases there must be a radio office at the airport, and operating personnel stationed there and also at the actual radio station.

The Air Navigation Division's radio engineers therefore suggested another course.

"We can't put the broadcasting station and the radio range beacons on the airport," they said, "but we can put the personnel there and let them use telephone circuits to operate the transmitters."

Rebate central equipment for this particular purpose was designed, built and tested. It was merely satisfactory. Now, at about airports the communication office of the Air Navigation Division houses the radio personnel, the telegrapher equipment, instruments for making weather observations, and remote control apparatus for the radio communication station. When it is time for a broadcast, the operator dials a number which starts the transmitter several miles away, picks up a microphone and reads the weather reports, lists another number and the transmitter goes off the air. Similarly, he controls the operation of the radio range beacons.

The split course problem

A difficulty which has not yielded so readily to our studies is that of multiple or split courses transmitted by radio range beacons in mountainous regions. The mountain, in some cases, cut upon the transmitted signals to make them appear as "no return" indications at points where pilots should hear "off course."

The regular radio range beacons have a range of about 100 miles. The pilot picks up a course, follows in 100 miles, passes over the station, and then follows the opposite course to a point about 100 miles beyond the station. If these courses pass over or between mountain ranges they are subject to the multiple course complication, which may easily cause serious trouble.

A radio engineer and pilot assigned by the Air Navigation Division to investigate this phenomenon recently tested an extensive survey which resulted in the recording of complete data on multiple courses, data which are now being coordinated and studied so that we will have a sound basis for our research. Several schools of engineers have already been invited to assign to graduate students research projects which will clarify the theory and principles involved.

In the meantime the bureau is introducing a new type of airway installation which will minimize this difficulty as well as others. This takes the form of an interrelated landing field with beacons and boundary lights, two-way radio

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communication facilities, and a discrete radio marker beacon at relatively short intervals along the airways. Courses transmitted by the radio range beacons to assist the more powerful radio range beacons, but can be received only for a distance of 25 to 30 miles from the station. Thus, with the directional stations at these intervals, the radio courses can follow more accurately any directional variations in the airway route. Also, difficulties from multiple or split courses need to disappear. Pilot courses rarely cross tracks within 30 or 40 miles of the transmitter, and if any multiple courses are set up by our new marker beacons, they are so close to the true course that the pilot is not led far from the airway.

This type of installation is being made on the Northern Transcontinental Airway, from Seattle to St. Paul-Minneapolis, now being constructed by the Bureau of Air Commerce under a Public Works Administration allotment.

Silence resolved

An important feature of radio range beacons directional products is the case of silence directly over each transmitter. Passing through the cone of silence, when he cannot see the ground, the pilot knows that he is directly over the radio beacon station, and, with one of the four courses he over the airport he can follow it to the landing area. But the cone of silence is comparatively small. An airplane flying at 125 mph,

at 1,000 ft altitude, passes through it in four seconds. At higher altitudes it is water-fort seconds for each 1,000 ft. Also, the necessary accuracy of the signals is a purely technical problem which may easily be controlled by the pilot, busy with controls and instruments.

To provide a positive indication of the



cone marked by a radio range beacon. During no return, the pilot knows he is directly over the station, and is therefore in the cone of silence. When he cannot see the ground, the pilot knows that he is directly over the radio beacon station, and, with one of the four courses he over the airport he can follow it to the landing area. But the cone of silence is comparatively small. An airplane flying at 125 mph,

at 1,000 ft altitude, passes through it in four seconds. At higher altitudes it is water-fort seconds for each 1,000 ft. Also, the necessary accuracy of the signals is a purely technical problem which may easily be controlled by the pilot, busy with controls and instruments.

To take all of the radio directional indications out of the pilot's headquarters, and put them into the instrument board, would be advantageous, in that the pilot's eyes would be available at all times to observe communications. The bureau has conducted experiments which indicate that this dual system similar to those now received easily by the pilot can be used to attract a needle on the instrument board.

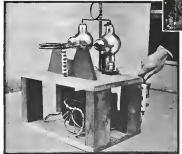
Instead of sending A, S, W and N (—) for the directional signals, the radio beacon transmitter is arranged that it will send a dot in lieu of the A, and a



Indicator for direct transmitters of sound needle rather than signals. The needle on all scales is indicated by the central position—the horizontal scale indicates values of signals (Ward).

dash in lieu of the N whenever the sound signals are to be converted into visual indications. In either case, the signals merge into a common scale, cut on-course. A converter in the airplane converts a needle in a meter on the instrument panel, meaning in no more, can use in response to a dot, to the other in response to a dash. When both signals are equal in strength, the needle points vertically to show ground.

Very many attractive features of this development are: (1) the pilot who does not have equipment to convert the sound radio beacons into visual indications, can listen with headphones, following the course in the manner that is now customary, and (2) the radio range beacons on the Federal Airways System could be equipped to send the dot and dash instead of A and N by a single



Experimental model of the marker beacon which transmits a signal to indicate the location of a radio range station. Based on the standard design a light is flash on the instrument panel.

clarity of signals at each station, with little expense.

Frequency selection

From visual interpretation of radar radio range beacon signals, the logical step is simultaneous transmission of radio range beacon signals and voice communications from one station on the same frequency.

Already communications and radio range beacon stations have been combined in some instances, so that both are based in one building and utilize the same antenna system, and the latter now is engaged in experiments with the objective of simultaneous transmission on a single frequency.

At present, the radio beacon signals are on the air continuously until they are on a weather broadcast. Stopping there while the weather reports are read, although it requires less than three minutes, may mean that step 40 of the air just at the time when an aircraft must urgently move. With simultaneous transmissions, there could go on the air at any time, and the radio beacon signals need not be selected.

It will require an extra antenna at the radio station, in the form of a fifth tower. As related beacons, radio range beacon signals now are broadcast from the four steel towers of the T. L. summit. The broadcast voice at the same time, on the same frequency, an additional tower, located in the center of the square, will be required.

Radio systems throughout the Federal Airways System now are being provided with fifth towers, but for a different purpose. At the present time, the fifth tower is used for noise transmission as an altitude frequency, when it is necessary to keep the radio beacon signals on an interference-free regular frequency. For example, one aircraft may need the radio beacon signals to lead it to the airport at a time when a weather broadcast is scheduled. In this case, an announcement is made on the station's regular frequency that the weather broadcast will be given on this fifth tower. The radio beacon signals then are resumed automatically, and the weather broadcast goes on the air from the fifth antenna tower.

Radio intercepts

I have already said that the bureau, in designing efficiency and safety of radio aids, looks at the same time to retain simplicity. Another research project goes along on how to improve efficiency behind the scenes, its objective being a selective response to radio signals. This project is the alternate substitution of radio for the present land-based intercept circuits.

Intercepting circuits are now used for transmission of weather reports, forecasts and maps, and reports on the progress of aircraft flying along the

airways. The leased wire circuit provides channels for passing this information along to support and to the radio broadcasting stations. On a few airways, point-to-point radio is used for collecting the weather reports from the various observation stations.

But point-to-point radio, as now in operation, has limitations that do not admit for development. The latter, even when fully automated, inevitably permits not a record of all messages passing along the airway. If the automatic transmitter could be harnessed to the radio wave, we should have the advantage of both. The bureau is now investigating various methods of transmitting by radio with automatic reproduction of the message as received, and if one of the systems is found practicable for the Federal Airways System, it doubtless will be adopted as a substitute for the present leased wire circuit.

Landings blind

Of all progressive advances in aeronautics radio, the one which will demand most from the pilot is radio landings, and which, at the same time, will do most to extend the usefulness of the airplane in blind flying.

A radio system for blind landing or aircraft which the Bureau of Air Commerce developed, and which consists of a radio frequency beacon, a landing beacon, and marker beacons to indicate the approach and the edge of the landing area, has a number of very desirable features. Numerous blind landings have been made with this system at College Park, Md. Newark Airport, and Oakland Airport. (See *Aeronautics*, July, 1941.)

The Bureau of Air Commerce has followed closely the development of a radio blind landing system by the Army Air Corps, and its working system is being used by the Army to determine the application of the system in civilian and commercial requirements and has entered upon the final phase of their work in this connection. They have already made more than 100 landed, instrument blind landings in instrumented airplanes since 1934. These records have come about only through the splendid cooperation of the Army Air Corps.

An interesting feature of the Army Air Corps blind landing system is the use of the radio engineers, or itself as cockpit receiving instrument. As a landing device it uses a portable radio receiver and earphone set which is light and compact and to which an interesting feature that he can receive.

The radio engineers between a standard form of equipment for air navigation, and at the same time a prime element in a blind landing system, one problem of blind landing technique is solved automatically. Under the present system, the pilot would become familiar with the radio aids at operation, and maintain his proficiency as he sits at a high

level. Thus when an emergency blind landing arises, it is ready for him.

After extensive tests at both systems, the Department has adopted the Air Corps system as the standard for procurement and construction.

Radio approach

For airports and landing fields at which communications do not justify installation of a complete blind landing system, the radio approach system developed by the Bureau of Air Commerce may be used. With it landings may be safely executed when ceilings are as low as 50 ft. This system can be provided at any field, even a radio range beacon transmitter, and no present practice is to direct one course of a radio may become over the adjacent landing area, that such additional system is a landing aid already exists. To implement this idea, the airport would have to provide two marker beacons, one on the approach to the airport, one of them 2 or 3 miles away, and the other over the edge of the field.

This system, installed above or in the clouds on the radio range course, would give the signal of the first marker and know that he should prepare for a landing. It would begin to land already, and by the time he reached the second marker, over the field, would be ready to glide on down to the runway. Tests of such a system conducted at Washington Airport, Washington, D. C., demonstrated that it is a dependable for landings with ceilings as low as 50 ft. A true blind landing in terms of means, is designed for landings with ceiling and visibility both zero.

The latter is a substantial radio currently will include blind landing. The radio compass also appears to be very desirable with the beacon and compass aid in working system. With the radio compass the pilot just above the clouds can point the nose of his airplane directly toward any radio beacon in use. Being adopted by airlines, Mechanical breakage of the set cover by a passenger device in panic moment by the air flow in the cockpit, or by other means, may cause disorientation of the set by test in use of keeping the air flow indicator glass under cover. Previous use of an automatic pilot in the cockpit and treatment render a place in popularization. There may be better still of functional importance. Here, for example, any radio aid that occurs the whole show. But at present the triple combination is feasible enough to justify confidence.

The present De-Long report, with its original development appears to Dr. William C. Gurn, formerly of Cornell University and current vice-president of the R. H. Goodrich Rubber Company. It was to him that funds were so given by the Goodrich Foundation for the construction of a refrigerated



The trend toward "no top" flying in bad winter weather has greatly stimulated interest and progress in the campaign to develop complete protection against the hazards of ice accumulation. AVIATION presents here its latest news on recent experiments and present practices.

THINK it is such a simple matter to fly in any kind of weather. A bill does pass you up when the prospect of ice accumulation on airplanes now first best studied these different methods were proposed. Upon that they were found to vary widely in effectiveness, but today all three find some application in the equipment being adopted by airlines. Mechanical breakage of the set cover by a passenger device in panic moment by the air flow in the cockpit, or by other means, may cause disorientation of the set by test in use of keeping the air flow indicator glass under cover. Previous use of an automatic pilot in the cockpit and treatment render a place in popularization. There may be better still of functional importance. Here, for example, any radio aid that occurs the whole show. But at present the triple combination is feasible enough to justify confidence.

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In the spring of 1932 a Northrop Alpha 4-A, was equipped with shoes on wings and tail surfaces and put into service on the eastern end of TWA's regular transcontinental run for four months. The last subsequent winter saw the entire Northrop fleet of TWA, De-Long equipped. This winter the Douglas transports are being added to the TWA provision list.

Other winter safety effort has been contributed by United Aerial long and active interest in the problem, it developed a De-Long aircraft for De-Long

247 transports and itself it exclusively through the winter of 1935-36. To its equipment must go a considerable credit for the development of the wind-tube tests, the wing-pole distribution valve system and other features of the most modern equipment. Last summer United placed an order for 30 sets of De-Longs to be used on Boeing transports.

TWA experiences

As an important part of its research program, TWA has engaged its pilots to fill out a questionnaire-report whenever de-icing equipment is used. Examining some 45 reports covering 1031 and the spring of 1939 gives one a remarkable picture of modern transport operations in bad winter weather. For a random sample—"Where were icing conditions encountered? Altitudes? Runways? Weather conditions? Snow, freezing rain, sleet conditions at surface? Top of clouds at 17,000 ft.—When was de-ice turned on? At 5,000 ft. over Albuquerque? How long, till snow, sleet, or rain stopped? How much snow?—Did not allow flying ceilings of ship 120 miles from the 750 miles north have been considered the equipment. Functioned satisfactorily." Many others lessening serious taking off with ceilings below 500 ft. with cloud tops at 15,000, or unknown in altitude. Rain, snow, sleet, sleet conditions in mountainous country, hundreds of miles of coastline encountered flying against headwinds, icy conditions, ice-accumulation. Hardly any of light snow would have been considered less than five feet high.

"Did not allow flying reports establish a number of general principles of De-Long practice. Wanted to get their air-transportation safety plan on their pilot's report. The plan was to get the pilot take beacons as soon as the plane approached potential icing conditions, radio. "Would not allow flying below 2 or 3 miles from the ground. The De-Longs two or three hours on end at 10,000 feet. The reduction of the landing gear down at 10,000 feet, up to 10,000 feet on lower. In addition, a landing of 25 to 30 seconds seems most effective. The winter months stand out clearly as the most critical period in approaching the end blind flying ceilings. Again and again the report and the De-Longs report on the winter months in understanding the light without De-Long even though it turned out that they were not absolutely satisfactory."

Of general effectiveness of the wing and tail surfaces De-Longs three equivalent data. "The ice created along the leading edges and along the wing and tail surfaces was a solid ridge along TWA's transport aircraft. That there was occasionally severe conditions under which no equipment, yet, almost no function, relatively, in the other

hand, definitely advanced, just as no one would deny that there are conditions to be encountered in thunderstorms, and neither work and logs that are all beyond the capabilities of present-day aircraft operations. To avoid such conditions airlines use every meteorological reporting and forecasting resource available, and pilots maintain the utmost vigilance. The spectacular loss of two mail planes over Pennsylvania last December as a single storm has been ascribed to a baffling and persistent failure in the observability of the air hauled by the entire air transport industry.

Weight reduced

The De-Loer has been generously refined in design details. The weight of the entire equipment has been reduced to 55 lb. for the Boeing installation, 75 lb. for the Boeing. The inflatable tubes built into the shoes are made of a specially treated neoprene fabric, coated with a high quality wax-like finish. There are welds on the leading edge of each outer wing panel of the Douglas or Boeing, four on each wing section between the fuselage and engine nacelle, two on each side of the stabilizer, and two on the front of the fin. Incidentally, the fin tubes are of great importance because of the marked effect on aerodynamics of the fin due to the effectiveness of the stabilizer.

Air is supplied from two Bellows pumps, one on each of the two engines, to an air separator filled with phosphor bronze granules. The cleaned air is then forced, at a constant pressure of 5 lb., to a distributor valve operated by a 120-volt motor through a 110-lb. collection tank with the structure of a bellows. This bellows leads the air to the De-Loer valve tubes installed in the skin of the plane. The use of a separate motor to drive the distributor valve represents a

distinct improvement over earlier systems in which the distributor was connected directly to the engine pump and varied in speed with the engine revolutions. With the separate motor drive, proper inflation timing is maintained under all conditions. The load on the pump is not great, one of the two valves being sufficient to operate the entire delivery system. To decrease driver burden, Bellows engineers have developed an in-flight engine air demand that act from an inflated set of tubes is discharged into a deflated set and the pressure allowed to equalize before the valve set is further reduced by the pump, thus substantially decreasing the amount of the pump motor load. The ordinary load is so small however that United Airlines experts have not deemed the device necessary for their present equipment.

Propeller guards

One of the primary problems of removing the accumulation along the leading edges of aerial sections was solved to the extent of permitting long flights as

far as-forming conditions, the effect of ice on the hub and roots of the propeller because of considerable importance. Ice-accumulation rates of rough inches were reported. Early designs consisted simply of spreading thick acetone oil on the affected parts at the beginning of each flight and at each landing point. Propeller hubs are now protected by large rubber-covered spinners and the roots by rubber covers. Specially absorbed, the rubber is then coated with oil. The device is reported as quite effective, though the oil coating still requires frequent renewal.

As to means the least important ice hazard during flights in bad winter weather is the plugging of carburetor and venturi tubes, with a resultant failure of the air speed indicator and all gyroscopic flight instruments at a time when they are most needed. A number of airline planes have therefore been equipped throughout the winter months with electrically heated static ports with elements between 4 and 5 ohms from the plane's regular electrical system. In contrast to the common use of heat in this way, it was found that the static ports also operated satisfactorily. Occasionally one is left turned on after the flight is finished and the engine is shut off. In a few cases mild ice accumulations have melted and the tubing and instrument casing is water. A few instances are reported of failure under extraordinarily severe icing conditions.

Venturi plugging has been tackled by the extremely effective method of replacing the venturi with engine-driven pumps, in some early cases supplemented by standby emergency connections with the engine manifold. The planes being equipped with De-Loers now connect their gyroscopic instruments to the intake side of the Bellows pump used to collect their pressure statics. Reducing jets and regulating valves to give the proper vacuum to each instrument set, of course, included in the system.



Feeding air passengers, always a problem, is rapidly becoming a major consideration as competition becomes more intense. The author who has worked the matter from both the airline and the research-consumer point of view, has been that free meals has hatches must give way to piping hot attractive meals served at the passengers' expense.

By William E. Berchold

DEK-DESK apple pie, economic bread broiled in butter and baked over-egg? Many passengers have been convinced with the power to desert passenger traffic from one railroad to another. Convinced as an event were important considerations among long-distance flying services, where ticket deliveries are rated second only to speed in determining passengers' choice of fares.

Airlines, however, have been too preoccupied with schedule changes, faster service and technical improvements that business of air catering has not been far beyond primary factors in their appeal to the air traveler. But the trend is on, and airline emergency departments are curious to play an increasingly important role in competition as increased between parallel services.

As soon as the service became seriously interested in building up passenger services, they were faced with the problem of providing facilities for feeding passengers. Because the problem was so closely linked with the inability to act swiftly, the company

and all arrangements for feeding passengers under the direction of Parker Bengtson, chief of transportation. Boeing Air Transport made a detailed study of food for the facilitation of air air passengers' that under the direction of a specialist in dietetics who served as chief dietitian at the time.

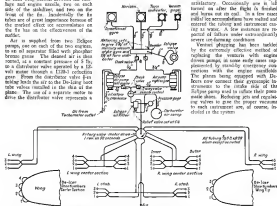
High-altitude dietetics

The conditions created by the different services were quite unique. Gravity, rock or lumpy foods were found to contribute to air sickness, if the passenger was in all unsteady in the cabin. The necessary a balanced diet was recognized. And, it was found advisable to serve relatively light portions at frequent intervals rather than to provide heavy three-meals-a-day service.

TAI took food that gold plates, lavender sugar, and gold plated knives, forks and other were provided in a reasonable technique to overcome boring passenger. Most of the food there, food, considered as either a high quality, light, and healthy, or composition plain, glasses and hotel menu which will not break off over much more than the light-colored. Flavors were equipped with foam food in six different bottles for hot and cold liquids such as coffee, lemonade, orange and apple juice, lemonade and water. All efforts to provide hot food or meat entrees failed, because of the excessive weight requirements for extra weight for extra weight and extra weight. The typical meal served aboard was standardized to include a selection of tomato or fruit puree, hot beefsteak or pork chops, bread, butter, and a hot sauce (usually chicken, cheese and ham), a fruit or vegetable salad, fruit, gel or cake, and coffee. Dinner menus varied little from those for breakfast.

When airline routes were based down to the level of mid-Pacific Island lines, the consumer departments were generally overworked to the operations departments with instructions to cut expenses. This was done as many lines by eliminating curtains, stewards, or hostesses and placing the task of serving meals in the hands of air-crew. The extension of the in-flight into the air, however, the departments were forced back into pure food lines, although one less diversified it served by making the no need to operations staff more than a few more staff. The problem of air catering is now solved at the time of the United Air Lines, Pan American Air ways, and a few special services of American Airlines.

As the problem of air catering is now solved on the basis of a discussion



Diagrammatic layout of De-Loer for Boeing 707-120 aircraft installation for a Boeing 707-120 aircraft. De-Loer valve tube in skin (1) and 1 and 1 in the stabilizer, passenger seats having been fitted directly around a De-Loer valve tube in the stabilizer. The entire installation weighs only 15 lb. and eliminates the weight and drag of direct vent systems.

RECENT enhancements of additional mail schedules by the Post Office Department, and its announced intention of making further additions, laid out some promise for improved carriage for various mail operators who have been suffering losses under the present contract. The real benefits, however, from the addition of new schedules are seldom appreciated, however, unless a thoroughgoing analysis of the basic costs of the various types of operations, especially of the relationship between direct flying costs and general overhead. Costs are not merely where costs are incurred where costs are dropped, or eliminated from consideration because costs are originally calculated on an basis of potential revenues. Mail operator analysis must include the costs that could have been flown without real loss with the load factors actually available. Each case that comes up must be treated on its own merits, but the methods employed in the following article may be generally followed for particular operation or any type of equipment.

Some lines given additional mail schedules have already been flying the same schedules with passengers and engines only, and since the cost of handling mail is slight, previously all additional mail revenue may be credited to cargo income. Except in cases where charges in schedules are required which reflect passenger revenues, or where actual passenger flightless passenger loads. It is believed, however, that such estimates are rare.

Thus in most cases where the extra schedule has not been flown prior to the authorization of the additional mail service, a favorable portion of the mail revenue will accrue to net income, for, in general, the mail segment, plus the additional passenger revenue developed, will exceed the real cost of flying the extra schedule. It is less available, the revenues being reduced to cover much, if any, of the general overhead or ground costs which normally would be allocated to the schedule, but the company's financial position will be improved by the direct loss of the revenue of revenue over direct costs. This should be appreciable where a strong carrying capacity is available, the equipment and fuel and ground crew necessary to handle the service and over-expansion of the new schedule. Where no additions to ground crew are to be hired, however, or where a new air plane cost is purchased to take care of the extra schedule, its cash costs, even though might be estimated that reflect the mail rate was quite favorable, and for the passenger revenue developed was



As new mail schedules are added to existing operations by the Post Office Department, air-line executives are faced with the complex problem of determining the real costs involved. To shed some light on this important subject, the operating costs of four modern transport airplanes have been analyzed in detail.

By *Alvin P. Adams*

National Airlines Corporation

substantial, the profitability of the additional service would be questionable.

Composition of costs

Recognizing, then, the element of management and its effect on the cost of any business, the operating costs of an airline depend on two factors: (1) the equipment employed, and (2) the number of airplanes flown. The first item influences directly the flying costs per mile, for the number of engines and fuel, and, however, have a great bearing on fuel consumption and engine overhead costs and airplane maintenance and overhead costs are dependent on airplane size and type of construction. The first cost of the plane affects depreciation and insurance costs, and finally, the speed of the equipment has an influence on practically all flying costs. The latter because of considerable importance in determining the per mile costs, were not flying costs of an airplane depend on the laws of physics, and, clearly, the more mileage that can be covered in an hour, the smaller the per mile costs. Insurance, as an airline's ground equipment, or overhead, increase but slightly with in-

creased scheduling, three costs will decrease per mile after it is proportion to the increase in schedule operated. Therefore, then, the real costs to be considered in calculating the return per mile would be reduced until they gradually approach the flying costs of the equipment operated.

Obviously, the flying costs of an airplane consist of flying per mile expenses, maintenance costs, and insurance costs. Maintenance of planes and engines, aircraft communications, flight and engine department, aircraft insurance, and aircraft insurance and expenses.

The overhead costs change little in the aggregate amount with increased scheduling, and therefore (on a per mile basis) vary greatly with the mileage flown. They include:

Depreciation.
Insurance.
Engine insurance and line personnel.
Communications.
Other miscellaneous overhead expenses.
Taxes and overheads.
General and administrative.

In the past there has been a definite tendency to consider the total flying cost to the cost of the additional schedule. As a result, some lines have been reluctant to fly additional services without payments in spite of the fact that such schedules might easily be covered on a developing a satisfactory amount of passenger business.

This assumption would normally be justified if flying equipment was being utilized in the slowest extent, and if maintenance personnel were constantly kept busy, in which case any increase in schedules would constitute a proper increase in expense, and, as ground and maintenance personnel have been far from the cost, however, for few lines are carrying such a load from 250-300 hours per month use of their equipment and schedules are actually so infrequent that there is usually not a sufficient amount of maintenance work to keep all maintenance personnel continuously busy. Actually, therefore, not only can the equipment itself do more work, but also the maintenance personnel. Where this is the case, the cost of flying extra schedules is substantially below actual direct flying costs in classified lines.

Although conditions vary in a different line, and, obviously, the cost proper presented herein will not apply to all cases, in general, it can be said that the comparison between the real costs of additional schedules (where no additional equipment is necessary) are:

1. Flying per mile expenses and maintenance.
2. Ground and air.
3. Motor expenses.
4. Motor overhead.
5. Motor and plane costs.
6. Passenger facility insurance.
7. Fuel facility and property damage insurance.
8. Widen seats construction.
9. Miscellaneous.

These, then, are the real costs to be considered in calculating the return per mile from extra schedules.

Flying costs

Flying Pilot.—Probably the largest single element of operating cost are the salaries and expenses of flying personnel. At present, most of the larger lines are paying according to a rate established by the National Labor Board.

Plane	Speed mph	Hour Rate		Hour Rate	
		Day Rate	Night Rate	Day Rate	Night Rate
740	140	\$12.00	\$14.00	\$12.00	\$14.00
740	140	\$12.00	\$14.00	\$12.00	\$14.00
740	140	\$12.00	\$14.00	\$12.00	\$14.00
740	140	\$12.00	\$14.00	\$12.00	\$14.00

The overhead costs change little in the aggregate amount with increased scheduling, and therefore (on a per mile basis) vary greatly with the mileage flown. They include:

The pilot's monthly pay check, then, is made up of all the base rate, plus the appropriate hourly figure, plus the mileage bonus of four miles per hour.

Average of the average pilot cost \$10 hours a month, the pilot is paid a base salary, using a rate of 10 percent per month on average compensation increase, and a figure of \$50 per month representing any-hour-base expense of the pilot and co-pilot. Table I represents the pilot costs of four modern twin-engine planes.

Fuel and depreciation

Fuel Costs.—Gasoline and oil constitute the second largest item of operating expense. Basic costs depend on (1) the price of gasoline and oil and the taxes levied thereon; (2) ground transportation; and (3) the speed of the plane. Although fuel costs vary considerably, it is left that a fair average price for the higher octane major gasoline including state and federal taxes, would result in 12 cents a gallon. A average price for oil would probably be 45 cents a gallon. Table II indicates a fair average fuel cost of each of the four transport airplanes.

Engine Depreciation.—Another important flying cost which is covered regardless of the schedule flown, is depreciation on engines and propellers. Although there is a great variation in rates used by different lines, it is believed that 3,000 hours would probably be a fair average. Some lines are using a 3,500 hourly rate while others use as high as \$300. Obviously propeller depreciation

must correspond with engine rates. These expenses will vary with engine and propeller costs, the rate used and the engine type.

Depreciation of airplane and accessories.—are not included here under cost of flying extra schedules, inasmuch as the depreciation on the airplane and accessories is not a variable cost. With few lines obtaining more than 120-180 hours of flying per month per plane, and where schedules and the costs will permit, it is quite possible to reach 250 to 300 hours per month (note line actually obtained 225 hours per month on the first of mid-century in the year of 1952) it is seen that considerably increased flying can be done before plane depreciation cost will become a factor in extra schedule operations.

Table III shows engine and propeller depreciation costs on the four planes.

Maintenance

Maintenance, Engine Overhaul.—One of the obvious maintenance costs which goes up with increased flying is for engine overhaul. When engine overhauls are given after a certain hourly period, it is clear that the airplane's speed will have a great bearing on the per mile cost. Other airlines will be the engine design, fuel shop efficiency, engine parts cost, and the individual line's parts replacement schedule.

A probable average of engine over-

Table I: Flying Expense

Plane	Speed mph	Hour Rate		Hour Rate	
		Day Rate	Night Rate	Day Rate	Night Rate
740	140	\$12.00	\$14.00	\$12.00	\$14.00
740	140	\$12.00	\$14.00	\$12.00	\$14.00
740	140	\$12.00	\$14.00	\$12.00	\$14.00
740	140	\$12.00	\$14.00	\$12.00	\$14.00

Table II: Fuel and Oil Costs

Plane	Speed mph	Hour Rate		Hour Rate	
		Day Rate	Night Rate	Day Rate	Night Rate
740	140	\$12.00	\$14.00	\$12.00	\$14.00
740	140	\$12.00	\$14.00	\$12.00	\$14.00
740	140	\$12.00	\$14.00	\$12.00	\$14.00
740	140	\$12.00	\$14.00	\$12.00	\$14.00

Table III: Engine Depreciation

Plane	Speed mph	Hour Rate		Hour Rate	
		Day Rate	Night Rate	Day Rate	Night Rate
740	140	\$12.00	\$14.00	\$12.00	\$14.00
740	140	\$12.00	\$14.00	\$12.00	\$14.00
740	140	\$12.00	\$14.00	\$12.00	\$14.00
740	140	\$12.00	\$14.00	\$12.00	\$14.00

Table IV: Engine Overhaul Costs

Plane	Speed mph	Hour Rate		Hour Rate	
		Day Rate	Night Rate	Day Rate	Night Rate
740	140	\$12.00	\$14.00	\$12.00	\$14.00
740	140	\$12.00	\$14.00	\$12.00	\$14.00
740	140	\$12.00	\$14.00	\$12.00	\$14.00
740	140	\$12.00	\$14.00	\$12.00	\$14.00

load times for three transports are given in Table IV.

Maneuvering distance increases as intended for plane transporters depending largely on the number of loadings or trips made, speed is not a factor in those cases. For example, such items as brakes, tires, cables, and landing gear, which constitute the bulk of these expenses, are replaced not on a basis of hours flown but on the actual condition of the part. This usually varies with the miles flown, or take-offs and landings, not the speed of the plane while in the air. As regards engine hours, when such things as spark plugs, etc., are replaced on a basis of hours flown, the speed of the plane is a factor. Although there is a wide variance in the figures of certain hours as regards this cost, it is believed that a fair average would be about 25 cents per mile, for the four airplanes used as exemplar herein.

Maintenance Labor. The amount of maintenance required depends not only on miles flown, or on take-offs and landings, but also on the hours flown. The inherent basis of the regular periodic maintenance checks. Naturally, maintenance features of the equipment and workload, sustained by the time it is viewed also affect this cost. The extent to which this expense is met on additional schedules, however, will vary greatly. Where the frequency of service is sufficiently great so that there is a steady amount of work moving through the maintenance department, it is quite likely that labor cost will increase proportionately to the amount of increased flying. This is usually not the case, however, and, for this reason, the maintenance crew can be counted on to handle somewhat more work with little or no increase in personnel. It is quite possible that on the average, additional schedules can be flown for about 60 percent of the labor maintenance cost per unit of the existing service. This is to say, where labor maintenance (exclusive of overhead) has been approximately 3 cents per mile, the additional schedule could be flown with a cost of only 2 cents per mile for this item. As a matter of fact, these figures actually represent a large savings labor maintenance cost for the four transports included in this study.

Maintenance, Fleet Overhead. No

additional cost is figured for this item as, with the new air mail transports now in use, it is felt that the ship will be depreciated before the overhead is actually needed. In this connection it is of interest to note that certain phase-out aircraft have between 5,000 and 6,000 hours without overhaul. It is hoped that new equipment will be able to exceed these figures substantially.

Table V summarizes the maintenance costs incurred from flying extra schedules for the group of various transports in applying take-off and material costs. Significant increases have been made for the rise of the phase and cost of parts, etc.

Insurance

Inasmuch as the present article assumes no increase in equipment needs, it is felt that certain increased schedules such important insurance items as crash, fire, theft, etc., are considered an overhead. A certain percentage of the first value of the equipment is ordinarily set up for these items, and the charge made monthly in proportion of mileage flown. This may be an economic method of charging for the insurance, since, with crash coverage particularly, the crash-based amounts would have been flying, but until the method is changed, the charge will not apply on additional schedule costs. There are two important insurance items, however, which are not in extra flying. These are (1) passenger liability, and (2) public liability and property damage. The latter two items, the former is usually sold per passenger-mile, and for the latter, sells per airplane-mile. Although rates on both increase very greatly, a four or five mile flight shows 3 to 4 percent increase for passenger liability, and a 5 to 6 percent increase for public liability and property damage. Assuming, therefore, the additional schedule flown resulted in four passengers carried, the additional insurance cost would be \$0.015 per mile, including public liability and property damage. This assumes a uniform insurance rating for all the planes and the fees operating there.

Considerations and miscellaneous.

Another important item entering into the cost of flying an additional schedule is a agency's flight commission. The present agency's commission approx-

imates 5 per cent, and, although the percentage of takeoff sold by agents will differ tremendously for various lines, it is likely that an average might show as high as 60 per cent of an airline's sales were on commission. Thus, if the extra schedule resulted in four or five additional revenue passengers, the agent would normally receive 5 per cent of 60 per cent of the total figure resulting at \$0.060 per mile, or approximately near \$0.060 per mile on the average.

Miscellaneous costs amount roughly to \$0.020 per mile direct expense of mileage. They include such items as additional telephone and telephone costs incurred by the extra schedule, passenger supplies and expenses, including maps, guides, meals, bus transportation to, from, and additional flight attendants or landing fees incurred.

Summary

All the cost elements for the four airplanes under consideration have been included in Table VI to produce the total direct flying costs.

Since one estimates show that the direct flying costs for the group of transports selected may range from about \$0.21 to \$0.33 per mile, it is obvious that relatively light revenue loads of additional passengers will make the addition of an extra schedule (under the circumstances outlined) economically justified. For example, an average group of 12-15 passengers carrying a load factor of only 25 per cent, the 4 or 5 passengers would yield a return of \$0.24 to \$0.30 per mile, or approximately enough to cover the direct flying costs. Any revenue from mail service increases directly to income. Conversely, if the mail volume is sufficient to meet a substantial part of the direct cost, a very few extra passengers will more than make up the total expense.

New schedules thus enabled may really prove to be traffic builders. By making extra services available to the public they may attract new customers. Before long, the land factors may increase sufficiently so that the use term "extra" becomes "regular" and the additional schedule is a proportionate share of the general overhead.

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OPERATION AT DESIRED CRUISING CONDITIONS, PART EIGHT

Climb and Descent in Cruising Control

By Edmund T. Allen and W. Bentley Orwald

THE "point-to-point" cruising speed of an airplane is not, perhaps, determined when the speed and the engine and surface characteristics are known in level flight. Climbing charts for the level conditions have been fairly limited in presenting altitude. They are available, but not sufficient. It is necessary, as well, to include the effects of climbing in, and descending from the speed arising altitude.

As engine power and supercharging increase, speed range and cruising altitudes also increase. Differences between speed in level flight and in climb or descent are greater, but longer time is consumed in getting up to and down from the existing altitude. If constant be assumed, therefore, then point-to-point flight at high altitudes is faster or slower than flight at the same power at low altitudes. Whether this time loss in climbing is, or saved in descending from a definite altitude is compensated for by the increased economy of altitude depends primarily upon the length of the trip, the altitude of the point of take-off and landing, and the magnitude and direction of the climb or descent. This factor will determine at what altitude it is most advantageous to fly. The manner of climbing and gliding also will contribute toward overall "trip" cruising speed.

In short, the "trip" cruising speed is the complete result of the entire operation from the point of take-off to the point of landing, depending upon the length of trip, the work encountered, and the speed during all portions of the trip.

Because engine reliability and maintenance costs depend on power output, it is usually desirable that the power be kept within the specified gross limit. It is recommended for climbing, however, that the output be maintained at the maximum allowable limit prior to the any lesser value. Excess climb margin for safety, especially at take-off, and furthermore, it is advisable to reach cruising altitude as rapidly as possible so that the more favorable air density and existing conditions can be established as soon as possible. It is generally preferable to keep the air well up in the interest of low manifold pressure, more

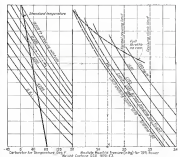


Fig. 10. Engine power varies the 10 per cent climb rate during the variation of table manifold pressure with altitude for various constant engine revolutions. Conversion note for reference: 1 horsepower is equal to 746 Watts. Approximate conversion to 1.34 hp per kw. For 1000 ft climb rate.

the probability of engine failure at constant power is greater for the lower air density, the manifold pressure and the manifold pressure itself.

It is recommended, therefore, that in climb the engine output be kept at the cruising limit and that the propeller be set as low as possible. Modifications may be necessary in certain cases, but unless an exception shows, that considerable gain results from operation of a different type, the low-speed, power-output limit should be maintained. This type of climb is especially favorable for cruising at the lower altitudes with higher supercharged engines, because of the rapid decrease in cruise velocity at the specified manifold pressure limitation.

Reference to the cruising curves in Fig. 2B (Airspeed, Density, RPM) will serve to illustrate the rapid increase in velocity below 4,000 ft, along the 30 in Hg. constant manifold pressure limit as altitude is increased.

Constant power curves

An estimated type of constant power engine curve is plotted in Fig. 3C, from which constant power can be maintained by controlling manifold pressure and engine rpm. An auxiliary scale is provided to facilitate the required correction for constant air temperature (which assumes power is inversely proportional to the square root of the absolute temperature). The use of this constant-power chart and mani-

Table F: Summary of Main

Plane	Engine	Miles	Yield
A	0.010	0.000	0.000
B	0.010	0.000	0.000
C	0.010	0.000	0.000
D	0.010	0.000	0.000

Table F: Total Direct Flying Costs

Plane	Speed	Altitude	Cost
A	0.010	0.000	0.000
B	0.010	0.000	0.000
C	0.010	0.000	0.000
D	0.010	0.000	0.000

any scale is similar to the use of the previous cranking charts. Several alternative types of chart for constant power can be drawn, such as shown in Fig. 34, or Fig. 7 (Aviation, May 1934) to which an altimeter air temperature correction scale would have to be added. Illustrative examples are included in Figs. 32 and 33.

The constant-engine-power curve is used principally for maintaining the desired power during the flight into an existing climb and descent. While the cranking charts for velocity assume level flight, the chart for cranking climb assumes constant power. The number of independent variables that can be conveniently handled in a chart is limited, hence the necessary (sometimes implied) condition for such type of flight (such as constant power, or level flight) makes the problem suitable for plotting control.

The constant-power chart can be used in any type of flight, whether climb, level flight, or glide. For example, should it be required to check the guaranteed cruising velocities at 75 per cent power at a particular density altitude, the pilot can readily obtain the necessary information by flying level at this altitude, and by adjusting the throttle according to Fig. 32 or 33 to obtain the manifold pressure and rpm required for the given altitude and airspeed or temperature. The actual adjustment

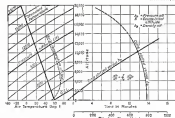


Fig. 34: Climb test results in a 75 per cent power, level-flight climb. Rate of climb vs. time at constant power. (Chicago DC-3).

is relatively easy at study of these figures will show. Since the intake manifold pressure has a greater effect upon power than rpm, the particular power is easily secured by setting the throttle for the appropriate manifold pressure, then correcting the manifold pressure slightly to adjust for the exact engine revolutions.

In a test where many factors tend

to cancel adjustment such as in the constant-power climb, it is advisable to assign to one observer the task of operating the throttle for maintaining the power by use of the power curves and engine instruments. In actual cranking operations, however, the climb and power are regulated indirectly through control of velocity and rpm or manifold pressure. In this way it is possible to control the cranking climb and glide in a manner consistent with the cranking charts for the level flight condition.

Climb test procedure

A complete flight test on climb at the cranking limit output of the engine includes climb at various speeds while the engine limit is maintained by means of the constant-power engine curves. Data are recorded at sufficient intervals of time to enable construction of an altitude versus time curve, from which is calculated the rate of climb. The following data should be recorded at approximately 1,000-ft. intervals except at limit:

1. Selected velocity.
2. Manifold pressure.
3. Pressure altitude (corrected at these intervals).
4. Time (correct at these intervals at use stopwatch).
5. Airspeed (when manifold pressure, engine revolutions, carburetor air temperature, and manifold weight (at beginning of each climb).

Additional data recommended

1. Exhaust back pressure.
2. Fuel and jet fuel injection.
3. Static temperature, etc.

The mathematical expression for the true geometrical rate of climb ($\frac{dh}{dt}$) in the ft-th sec system is,

$$\frac{dh}{dt} = \frac{dh}{ds} \cdot \frac{ds}{dt} \text{ per sec.}$$

$$\frac{dh}{dt} = \frac{dh}{ds} \left(\frac{ds}{dt} \right) \text{ (ft/sec)}$$

The notation is standard. A simplified expression which is convenient because of the use of the direct altimeter reading gives,

$$\frac{dh}{dt} = \frac{dh}{ds} \cdot \frac{ds}{dt} \text{ (when } h_s = \text{ pressure altitude)}$$

This relation follows from the fact that the change in pressure, hence pressure altitude, is related to vertical proportion to the air density or air temperature. The ratio of the altitude air temperature to standard is often sufficiently close to constant through a climb to permit the use of a fixed value of $\frac{T}{T_s}$ in the reduction of the climb. For $\frac{T}{T_s}$ is constant throughout a climb

$$\frac{dh}{dt} = \frac{dh}{ds} \cdot \frac{ds}{dt} \text{ (when } h_s = \text{ density altitude)}$$

This relation is approximately correct when no temperature variation is present; $\frac{dh}{ds}$ and $\frac{ds}{dt}$ will differ when the variation from standard temperature changes with altitude.

Rate of climb curves

The true rate of climb in measured distance at constant engine power and reduced velocity depends upon density altitude, hence is plotted against density altitude as shown in Fig. 35, where the flight test results for the altimeter rate

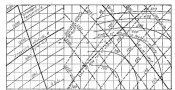


Fig. 35: Cranking climb chart for 75 per cent power, showing the velocity versus density altitude as a function of engine revolutions, indicated velocity and rate of climb. The chart is constructed by repeating a constant velocity with engine revolutions. The above curve has been estimated from one test to illustrate the general type to be expected. (Chicago DC-3, two-throttle propeller altimeter).

test results at 14,000 ft. pressure altitude and 50 mph. The pressure altitude is 75 per cent power climb is obtained at 100 mph for flight at 10,000 ft. (density altitude).

of cranking climb are plotted. Likewise rpm and indicated velocity vary with density altitude. The true geometrical rate of climb at constant engine power can therefore be plotted against density altitude for fixed values of the various related characteristics. Reduced results are shown in Fig. 35. Fig. 36 shows the general form of chart for the cranking climb which can be constructed from a thorough test of 75 per cent power climb in various reductions. The forms of the constant rpm and constant rate of climb curves have been estimated to illustrate the general relationships between the rpm, indicated velocity, and rate of climb. Fig. 36 is recommended for plotting control of cranking climb.

Controlled descent

The descent into the terminal point on the scheduled run should be carefully controlled. This descent is at least longer duration than the climb because of the general experience that economy rather than speed is desired in descent greater than about 400 ft. per minute. A cranking power glide from 14,000 ft. will require a minimum of 35 minutes and will cover about 120 miles, while the 75 per cent power maximum climb will require only about 22 minutes. It is essential then that the descent be controlled as a part of the cranking flight, and that cranking power thrust be maintained in its entire portions of the flight.

The controlled descent converts what is otherwise a necessary penalty at the end of a flight into a means of recovering in a large measure the energy lost while climbing. The rate of change in potential or "height" energy into kinetic energy tends to offset desired thrust horsepower for propelling the airplane. It is of interest to note that the

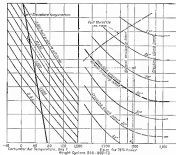


Fig. 36: Engines power curve for 75 per cent power, showing the velocity versus density altitude as a function of engine revolutions, indicated velocity and rate of climb. The chart is constructed by repeating a constant velocity with engine revolutions. The above curve has been estimated from one test to illustrate the general type to be expected. (Chicago DC-3, two-throttle propeller altimeter).

Example: For 14,000 ft. pressure altitude, 50 mph. The pressure altitude is 75 per cent power climb is obtained at 100 mph for flight at 10,000 ft. (density altitude).

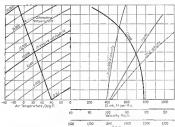


Fig. 37: Curves for the control of maximum climbing effect at 75 per cent power. The chart is constructed by repeating a constant velocity with engine revolutions.

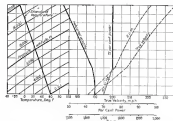


Fig. 37: Curves for constant maximum descent rates as you see altitude rate of descent. *Figure is not to high precision. (Charts obtained from Flight Test on Douglas DC-45)

400 ft per minute maximum descent corresponds 215 thrust horsepower, or about 272 effective brake horsepower, so the two-engine transport for which charts are being prepared. This power amounts to approximately 19 per cent of the rated engine power. In a 75 per cent power glide at 400 ft per minute descent, the airplane will develop the speed normally achieved at level flight at 94 per cent of the rated maximum power. Thus at 14,500 ft, the normal velocity of 265 m.p.h., will increase to about 325 m.p.h., while 191 m.p.h. at 8,500 ft increases to 210 m.p.h. This considerable increase in velocity as the distances of the descent lengthen, compensates for the loss in velocity during the climb. Hence for trips of any considerable length it will be found that there is an appreciable but not large difference between the overall "top" cruising velocity and the level flight cruising velocity at the same power.

The climbing descent is of course carried out with high pitch propeller setting. It is generally recommended that the rate of descent be kept as close to the 400 ft per minute level as possible, although a larger, flatter angle of descent may be found desirable in certain cases.

Descent tests

The characteristics in the descent are obtained through flight tests in much the same manner as the data on climb are obtained. In the test the approximate rate of descent is maintained by means of a rate-of-climb indicator and the power is maintained by means of regulation of the throttle setting to the constant power curves of Fig. 32 or 33. The same data are recorded as previously noted for the climb test.

The data are reduced through use of the rate-of-climb equations since the

descent is obviously only a negative climb in principle. Hence the natural rate of descent and the corresponding curve characteristics for the given constant power depend upon air density, and may be plotted against density altitude.

The results of flight tests on a 75 per cent power cruising glide are plotted in Fig. 37. It appears most convenient to maintain the same cruising power in the descent as in level flight, there is no provision for maintaining maximum cruising power in the climb. It is believed that the 400 ft per minute rate of descent should be accepted as the general criterion for all

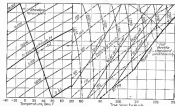


Fig. 38: Graphing chart for 400 ft per minute descent showing velocity versus density altitude as a function of per cent power indicated velocity and engine revolutions. The descent is controlled by indicated velocity and engine rpm. The curves have been indicated thus and tend to illustrate the general type to be expected. (Obtainable DC-45; high pitch propeller setting)

Example of present altitude of 13,000 ft, velocity and 1410 rpm. The percentage of rated power is then 75. Maximum points and desired 3000 show example.

glides. A complete test on the existing power glide for existing conditions would then plot the velocity and rpm for various percentages of engine power and 400 ft per minute descent. Fig. 38 illustrates the general nature of such complete tests on the existing glide of which Fig. 37 shows the results for 75 per cent power only. The similarity of Fig. 38 to the level flight cruising chart is worthy of note. Fig. 38 is recommended for plotting control of the descent in cruising operation.

The charts for the existing descent and climb which have been developed can be used by the pilot for replacing his entire cruising glide. The desired power and rate of climb and descent are indicated by adjusting the engine rpm and the indicated airspeed. In the level flight portion of the trip other indicated airspeed or rpm is sufficient to define the power and speed. Since the descent should be used for setting all of these characteristics, the density altitude conversion scale is indicated on all charts. For the conditions that level cruising operation, it follows that the density altitude and factors which comprise density altitude are most amenable for control.

It should be noted that appreciable changes in weight, propeller setting, etc., will require modification of the preceding results, which modification can be obtained either through calculation or subsequent flight tests. Small changes can usually be neglected.

For cruising operation and the establishment of schedules, the time consumed in each portion of flight must be considered. The overall scheduled time can be estimated from charts now developed for climbing, level, and descending flight.



Mac and Score

MacRobertson Score Sheet

Post mortem material for
a winter's debating

THERE ARE a number of syndicates by which the squares of an international air race might be measured. Its monetary cost might be weighed against its value in its stimulus to technical progress. The effect of its small scale reward on the general public's attitude toward air travel might be estimated. There are factors of sponsorship and management and the size and importance of the entry list that might be discussed. But by and large, the real test of greatness for any international contest will always be the number of competing nations which feel satisfied at its outcome. And the MacRobertson race for all its record breaking, accomplished nothing more commendable than its debatable major triumphs to so few that these competing countries, save two more than any comparable international one, had ever seen leave in such appreciable satisfaction. To Britain, none, first place and an unopposed racing victory. To American designers an aerial of triumph for their ingenuity in the transport field. To the Dutch, a world-wide notice of their bid for top rating for their route from Europe to the Far East. Dutch birds were frustrated and in these still remained a chance for a spectacular moral victory.

But if none of the major prizes were pretty well worked out, a lot of minor ones remain to see the scheduled of all nations. All the French, Italian, and German entries were missing, and there were enough actual American, English and American to cut the dead numbers that turned up at Woburn to 11 from an original entry list of 64. What could Wiley Post have accomplished as his unopposed ship? What could Harold Gatty have done with his expedition of the route? Would Gatty be too Douglas entry have broken the Coast? Then there is that delicate problem of the race committee's subject to the President. Referring to the full allowance of the Douglas and some others. "Who the committee seeing far and safety in cutting down reserve, range to course will lead take-off and landing safety? Douglas values are also all well-proven forces.

As to the actual contestants, what could the K.L.M. Douglas have achieved in a full on race? Placing second in such a contest while carrying three passengers, however, would add 20,000 letters and making regular stops along the K.L.M. air route to Hawaii is one of the most outstanding demonstrations of any transport performance on record. Incidentally, he should not mistake the Dallas-based Comet and its company. One Comet entry was. Another gained a poor fourth. The third was forced out in Karachi. All of them had engine trouble that day so did the majority of the rest of the contestants come through along the course.

Alas, too, we have heard from an expert on navigation on the subject of some of the off-course wanderings of most of the leading contestants. The boats of Scott and Shackleton and others were known to land short of Baghdad. Turner and Pughen's personal misadventure



Douglas and Mac

TWO



Douglas and Mac

THREE

and their maintenance are one of the most outstanding factors in the present aviation service. Government on the feasibility of establishing temporary air mail routes, has exposed the opinion that aircraft already developed in this country are suited to such service, and that operations could be inaugurated within a few months. He urged the adoption of a government program that would give training to the industry for a period of years and predicted a tremendous increase in airline activity as such a program was forthcoming. He also favored the establishment of a permanent commission offering centralized control over commercial aviation.

Government appropriations for experimental work on developing improved planes as a means of stimulating private flying were recommended by representatives of the Private Flyers' Association. A national program for the improvement of present landing field facilities was also urged. The association was member of the group to recommend a permanent commission.

Other well-known board-level General William Mitchell, commander of American Air Forces, director of the Military Code, Edward V. Rickenbacker, World War ace; W. B. Grigg, Chief of the U. S. Weather Bureau, Stanley C. Kennedy, of Trans-World Airways, James Clark Thomas E. Doe, vice-president of Sperry Corporation, and former president of Eastern Air Transport, I. I. Schmitt, Schmitt Aviation Company, Commander Frank Hawks, World War hero; Dr. Karl Goetz, president of the Massachusetts Institute of Technology.

GHQ Air Force

Last month President Roosevelt approved the War Board report authorizing an ultimate goal of 1,500 biplanes annually for the Air Corps. With Presidential approval, the War Department proceeded to carry out the War Board recommendations. Most important change in the Air Corps General Headquarters. As the main tactical element of the country's aerial defense, the GHQ Force will consist of six to ten 1,000 bombing, pursuit and attack planes in five wings, based at Army fields on the Atlantic and Pacific Coasts and in the Middle West. Headquarters will be at Langley field, Virginia. Headquarters of the Air Corps, including observation squadrons, remain under the command of the Corps area commanders. Under the new arrangement, Chief of Air Corps Facilities will be in charge of most of Corps flying activities. He continues to have charge of schools, training and procurement as well as the War Board report which stated that the pri-

mary function of the Chief of Air Corps should be to direct efficiently the business and technical side of the Air Corps development and procurement problems. No one has as yet been named to command of the future chief General MacKenzie.

To check up on the efficiency of Air Corps personnel, the War Department ordered tests to determine the flying ability of all officers in 43 more days. Officers' personnel. All generals, colonels, majors and a majority of the captains will be examined. Those disqualified from further flying will be assigned to ground duty, transferred to other branches of the service, or retired. There are 3 general, 9 colonels, 38 lieutenant colonels, 18 majors, and 362 captains in the Air Corps. Elimination of any found unfit for flying would make way for the promotion of new and junior staff.

Ground school training and barracks hours of an experience will be given to all First Point cadets of the first and third classes at Maxwell Field, Ala. In order that the maximum number of pilots may participate in each flight and limit varied training in navigation, bombing, gunning and observation, the flights will be carried out in bands. Staff Secretary Davis is sponsoring the new training. "It apparently hope to complete flying of West Point as accomplished air pilot."

To improve the condition of war training, the War Department has changed the composition of the Air Corps Board which formulates aviation technical standards for all Air Corps units. Personnel members will be reduced from five or more than eight.

will be designated by the Chief of Air Corps and will give full time to their duties in connection with the board. In the past, the work of the board has been done by officers performing primary duties as instructors at the Air Corps Technical School. Its officers members of the board will be, as formerly, the commandant and the assistant commandant of the Air Corps Technical School.

Light plane contract award

Hammond Aircraft Corporation of Ypsilanti, Mich., was the winner of the competition sponsored by the Bureau of Air Commerce to produce equipment for its inspectors, and at the same time to encourage the development of an economical, safe airplane for the private owner. The contract awarded Oct. 18 is a contract for delivery of the Hammond planes (see Flying Equipment, page 37), for 140 technical details, with the option of taking five more, at \$3,750 apiece. Delivery of the first plane must be made within five months, and delivery of the remainder within five months after the Bureau's acceptance of the first. The Bureau originally planned to purchase 25 planes, from the winner of the competitive bidding, and ordered the number so that a portion of the work set aside for the purpose will be available for experimentation with designs submitted by other manufacturers.

F.A.A. conference

Delegates to the 24th convention of the Federation Aeronautique Internationale, the governing body for world aviation, met in Washington, Oct. 6-11 in connection with the National Aeronautics Association, no U.S. officials. Leaders from



NACA HOSTS TO F.A.I.

After the F.A.I. Kite Race, some of the "World's" most experienced members of the 14th congress of the Federation Aeronautique Internationale spent the night in commercial equipment in Langley Field.

22 of the 34 national clubs represented on the Federation were present, including Prime Minister of Romania, its president, Paul Tudosescu of France, its secretary-general, and Louis Blumenthal, United States delegates were Gustaf L. Collet, Herman Buchner, F. Tréhen, Debraux, Edward P. Warner, Chicago.

In closing regulations were promulgated to promote the development of private flying throughout the world. Until the present of London it was decided that each country would prepare a map for its official Air Guide showing the exact position of airports and published a report urged that as much territory as possible be opened up for air touring, pointing out that it has already removed all such restrictions except for the Swiss Club. Negotiations maintained a limitation of regulations on the use of cameras and radios in private aircraft.

International regulations concerning trials for accidents were also revised. Speed records will be made hand-drawn by electro-photographic methods with a tolerance of not more than 1/10 second. Mutual credits for a new speed record will be allowed only if the existing record is bettered by at least 5 mph. For altitude records no change in rules was considered so require the carrying of an automatic pressure-sensitive fuselage recording unit, instead of the simple barograph now specified.

As proposed by Egypt last year, the Federation was granted official recognition from the League of Nations as a non-official authority on aviation relating to international private flying.

Recent arrangements, the Hungarian Aero Club was admitted to membership and Prince Billewicz was announced re-

electoral president. The next Congress will be held in Belgrade, Yugoslavia in October, 1935.

[We are indebted to a large part for the above material to Gregory A. Smith, secretary of the Department of Aviation in England and director from its country to the F.A.I. Conference.—H.]

Wallops heads V.A.C.

In the annual meeting of the National Aeronautics Association Oct. 11-12, Senator William G. Wallage, who has long been an enthusiastic plane owner, was elected president. He succeeds former Senator Thomas Bagshaw, holder of the office for the past two years. Major James H. Doolittle was elected vice-president and Mrs. Louise Thomas secretary. John F. Votey was re-elected treasurer. The following governors at large were chosen: Dr. W. H. Wright, Griffith, John Porter, William H. Wright, William H. Wright, Jr. and Frank Hughes. A resolution was adopted naming Dr. W. H. Wright to reduce to six months the plan in which he and his brother made the first successful flight in Kitty Hawk, and which is now in the New Kensington Museum in London. A committee of four was appointed to confer with Dr. Wright, including Edward P. Warner, chairman, Frank H. Hughes, Dr. George M. Lewis, Paul S. Phillips, Capt. William J. Lutz, in whose honor the Wrights' flight while they were in prison during their flight.

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New planes for old

Aviation progress toward early adoption of water balloons as airplanes for low speed sports are being. On the West Coast, American Air-

line delivery on the first of six Douglas transports is in place in service within the next few months. Its 1,000-600 is a significant progress made, besides the Doolittle, the Curtiss Condor and the Vultures, some of which are already in service. The six planes are replacing all of the company's other three-engine transport. Six of the new Condors are the lightweight single plane, 1,000-600, income passenger type, some carried by American Airlines, are others now operating between Los Angeles and Dallas and between Chicago and New York since the service were inaugurated in the early summer.

Eastern Air Lines will put two more eight seat Douglas planes by the beginning of next year. In addition, the Curtiss Condors will be added to give higher speeds.

Recent shipping flights marked the delivery of the first of six Douglas transports to Pan American Airways. One of the three intended for Panagra's service between Lima and Shanghai, China, was delivered Sept. 12. The plane made the 1,700 mile flight from Cleveland, O., to Lima in two hours, 45 minutes. The previous record was fourteen hours. Another new Pan American Douglas, the San Martin, set two hours, 25 minutes from the Santiago to Buenos Aires on the first flight over the trans-Andean mail route Oct. 11. Madrid now was four hours, 35 minutes.

Further passenger service from Philadelphia to points in the Middle West and Pacific Coast resulted from approved schedule changes by United Air Lines. Two from Philadelphia to Detroit via



F.A.I. IN EXECUTIVE SESSION

Robert Anderson appears before the President's air policy board. Members of the Commission from left to right are: General C. E. Jones, Jr., Colonel Edgar Harold Edwards, F. Warner, Albert J. Bowers, and J. Stewart Davis, secretary, in standing.

SIDE SLIPS

By Robert R. Orborn

WHEN SEEN by the papers that one of the competitors in the England-Australia Derby was expected to land at an intermediate point "where fresh fuel will be waiting for him." His aviation now reached the point where we are going to have to have "fresh" fuel in order to "eliminate the harsh irritants" from our smokers? Heaven help us if we are going to begin to "look for the date on the can" in order to prevent neurological habits, pink tooth brush, or R.D.

When asking the airlines, that very uncomfortable and dignity person you may notice to one of the passengers is very probably one of the flight pilots dead-heading it back to his home port. While the rest of the passengers are "belching" or trying to get the attention of the hostess, this nervous passenger frequently disapproves of each take-off, and grinds the axes of his seat for every landing. Even on the new night sleepers, he cannot rest in comfort—a pilot friend of ours told us that on a recent trip on one of the new Pullmans of the air,



couldn't sleep a wink all night because the pilot in charge didn't properly synchronize his engines. The rest of the passengers, of course, had a good night's sleep.

Breakdown of the new sleepers, reached us of some extent we have been thinking to pass down in the accompanying public. On our trip of the sleepers recently, just before a heading the pilot came to the conclusion that he was not going to be able to get his heading gear down and was going to have to make a where-ye heading. So all passengers were advised to get up and get on safety belts, which they did. After they were all set, the hitch in the retracting mechanism was corrected and everybody went back to sleep. So, if you have a weakness for loud and brilliant colors in your night dress, or if persistence you are in the conditions you reported to depress with night dress, you had better have a spare and changed pair of pajamas.

available under your pillow,—the land-
ing gear might not work.

According to an article in the New York Times, a new type of passenger transport airplane is equipped with "flaps which permit the reduction of wing area." Bob Perry comments that he has heard of this method of reducing wing area before, but the trouble in the past has been that it works only once.

[illegible]

It seems to be a rather questionable practice of the airlines to be promoting the development of "luxury" air transport ships in the middle of a depression like the present one. As another example of the emphasis being placed on luxury we have a recent specification put out by a prominent airline for a new transport which states "High Life devices to be provided."

The October issue of *The Aviation Forum* has several articles pertaining to aviation, one of which describes the new city airport for New Orleans. Note that the "luxury" smile of the aviation business is again

AVIATION
November 1986

stressed: "The passenger accommodations comprise a two-story waiting room with mainline baggage claim and check-in counters; a lounge; a dining room; restrooms; lavatories; men's and women's lavatories; smoking lounge; and station and ticket offices. All are very elaborately decorated and furnished. At first light it almost seems as if the luxury scene had been overdone, but a second thought realizes that while one can put up with a brief pain when boarding a luxurious liner the baroque is tolerable only as a view of the comfort to come. The best airplane to date seemed to us to furnish no great degree of comfort as a ship and a plane. The air terminal facilities should make up for this lack as much as possible."

We had always thought that the reason the runway companies had been paid for terminals was so that the customs inspectors would have room in which to spread around the contents of our trunks and bags.

We are very much disturbed by the disruption of Mr. Wiley Post's new flying suit, suitable for the 40,000 ft. altitudes at which he now plans to operate his "Winged Man." His suit is composed of rubber sheathing with a dorsal helmet and chord harness. The suit is made of a material similar to paper with pipe threads around the joints. This spotty ensemble most closely resembles something Dr. Wilbur Wright might use in his search for new biplane hubs, and it enables Mr. Post to represent himself as still in the infancy of aviation.

The suit is also a very poor hazard suit. The suit is made of a material similar to paper with pipe threads around the joints. This spotty ensemble most closely resembles something Dr. Wilbur Wright might use in his search for new biplane hubs, and it enables Mr. Post to represent himself as still in the infancy of aviation.

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edly we shall have the student pilot
clanking about the fields in 100 years or
more to come.

Even at that, we'll have to admit that the new type of high altitude flying suit is far better looking than we had expected it to be. From first descriptions published some time ago we had expected Mr. Post to appear in an outfit somewhat like that of The Wizard of Oz's.

FLYING EQUIPMENT

Contract Winner

WHILE the Bureau of Air Commerce has sent out a call last spring for bids on 28 airplanes for the use of its inspectors and officials, the accompanying specifications provided more information among American aeronautical engineers than anything at the time. The specifications were prepared by the Civilian Aircraft Corporation, Inc., the Great Lakes Aircraft Co., and the Goodyear Aircraft Corp. as members of the "Goodyear Safe Aircraft Competition," held a dozen years ago. On Aug. 29 bids which had been received from fourteen companies, were opened in Washington.

The contract was awarded to the Goodyear Aircraft Corp. because it offered the lowest bid—\$100,000. It included 15 of the 28 to the Hammond Aircraft Company of Ypsilanti, Mich. To say that interest among designers and manufacturers has waned at a high price, would be pre-

that preliminary details of the winning design are available, for the Model Y aluminum will be a definite departure from the orthodox along the path approved by the Stout-Snyder and the West-Whitcomb teams.

A low-entry, three-wheeled, cabin compartment pusher, there are no important features of its design which are complicated or not previously tried in other designs. But an effort to refine the performance, construction, and production possibilities of such a ship is an interesting and worthwhile program of design research as the bureau could have selected.

are carried on two booms (spars) extending back from the wing structure.

Provision is made in dimensions and weight schedules for 40 lb. of baggage and a parachute for each occupant. Excellent forward visibility, a low nose level due to the engine position and a smooth airflow around the nose, and a door close enough to the ground to be really convenient guarantee an exceptionally attractive cabin accommodation.

Power is furnished by a Memac four-cylinder in-line inverted engine which will be equipped with starter generator and liquid storage battery. Special attention has been given throughout the ship to oil and grease fittings, replaceable bearings, inspection openings, and the general reduction of maintenance costs.

From the three-wheel feature and the general arrangement and proportion of the wing and control surfaces a number of operating advantages are looked for. In taking off the control stick can be pushed completely forward allowing the ship to gather speed. Then if the pilot should see that he cannot clear some obstacle he can simply lift his seat and make a short stop without moving over. It will, moreover, be impossible to stall the plane on take-off even with the stick full back.

In hoisting, the gliding angle can be made very steep if it is desired to hoist in a small space. The ship can be hoisted also by merely pulling the stock out back. For this latter case the forward and vertical speeds are designed to be very low and the landing gear is provided with ample shock absorbing capacity. Still hoists also have carefully guarded against for this type of hoisting. In general, landings will be possible at any speed up to that of ordinary cranes. Once on the ground the plate should be supported from the usual hazards of wind, rain, frost, snow, etc.

ing, and the effect of varying gain.

Last July *AVARON* published in it the elaborate specifications which have controlled the design of the Model Y. For completion of its contract the Blumson Company must demonstrate within four months that its plant has a 100% yield rate, that it can produce and ship a 25-ft. diameter bowl at standing speed, 600 in. diameter and less than 400 lb., over an altitude of the same height. In addition it must possess a design or two exceptional safety features of central and emergency, safety fail-outstrut equipment and fail for 8000 g's and be a paragon of detail design, assembly or operating and maintenance sources.

Genesis to Begin Productions

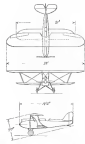
OF the most hoped present signs of renewed activity in the printing market is the continuing increase in the number of new entrants to the industry, for sometimes feared too late to resist, they have returns to the battle. The announcement that Michael Grogan has joined the Gropius Associates Corporation, leased shop space at Roosevelt Field, and is starting production on five glassers is especially encouraging. The place for the designer of the Ried, for the stress analysis of the Beverly Hills airport, for an architectural campaign of 23 years experience is in the front rank of the industry not outside it.

The plane the new ship will produce has been designed specifically for the lowest practical price range, combines with sturdiness and adequate performance. In many ways it is a scaling down of its bigger predecessor the Breda two-place open biplane, powered with the 80 hp. Curtiss or Wright Gypsy, is superior with its ease and simplicity.

The footage is of welded mild steel.



Wired command model of the Movement Model V of which Shoen has been asked to be the Master of the Foremaster subject in various mountainous tasks. A low wing two year stable position is will be occurred by the one cellular invited Nations. From the three



The design has three sporty and distinct. Marking the 28-70, Wright Bros. or Cessna it has been specially designed for low priced production.



The landing gear of the 28-70 is a simple design. The main wheel is a 12" x 12" x 12" oleo-pneumatic strut. The oleo-pneumatic strut is a 12" x 12" x 12" oleo-pneumatic strut. The oleo-pneumatic strut is a 12" x 12" x 12" oleo-pneumatic strut.

Spring and landing gear are in a single place with the front lower and rear upper parts. The entire section, supported by carbon steel struts, contains the main fuel tank of 20 gal. A variable-reverse rack holding 12 and 14 and fixed with a single foot type gear is mounted in the trailing between the engine and front engine.

The ailerons and elevators are connected to the dual control sticks through links and cable mechanisms, ailerons being located in the upper wing only. The rudder is connected to the foot pedals



The tail section of the 28-70 consists of a 12" x 12" x 12" oleo-pneumatic strut. The oleo-pneumatic strut is a 12" x 12" x 12" oleo-pneumatic strut. The oleo-pneumatic strut is a 12" x 12" x 12" oleo-pneumatic strut.

through the main cables. The stabilizer is adjustable.

Working at 800 ft. empty, 1,830 lb. fully loaded, the plane produces exceptional reserve power. It starts with full load 1,000 ft. in the first minute, has a top speed of 110 m.p.h. automatically, no extra conservative landing speed and control near the stall.

Bellanca Sweep

THE AMERICANS one of the biggest disappointments of the MacRobertson Race was the forced withdrawal before the start of Col. Pomeroy's Bellanca monoplane. Acquisition of the price and cost of the making which caused the withdrawal, the slower of one of the most favorable American designs was a delicate and unprofitable loss to the contest's value.

Designed from the outset for the

MacRobertson race, the Bellanca Model 28-70 has on test reached a top speed of 110 m.p.h. and has taken off with a gross weight of 8,000 lb., carrying fuel for a non-stop flight of 2,500 miles at a cruising speed of 100 m.p.h. Its 14-cylinder, four row, three cylinder engine develops 200 hp at 8,000 ft. and weighs a two-bladed Hamilton-Standard convertible propeller in 10 ft. 6 in. diameter.

In their basic layout, the Bellanca designers set themselves the task of retaining the advantage of a thin wing currently known monoplanes and a retractable landing gear. Their solution takes the form of a wing loaded by the main to the top suspension and to two steel supporting struts extending below the fuselage, the spacing of the lower bracing being such as to allow clearance for the landing gear wheels as they are retracted into the wing root. The wing is tapered in plan form and tapered in thickness near the tip.

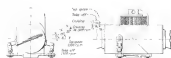
The landing gear of choice monoplanes must add the same weight to give it a fully developed, streamlined form. The center of the two cylinders, extending from the fuselage to the wing, is quickly detachable in flight from either cockpit. Two cylindrical tanks, of 200 gal. each, occupy the entire cross-section of the fuselage, being designed with special channels through which the fuel is drawn. The engine is supported by a full NACA cowling and 23 ribs, between the fuselage and other parts of the plane have covered the almost of control design to maintain clear.

The landing gear while retracting without completely while covered in the wing which are covered with flaps when the gear is in the up position is a retracting by means of a positive drive and cable mechanism and works in this short shorter unit. Brakes are of standard Bendix design, and are actuated by a hydraulic system. The wheels mount 24-in. tires. The tail wheel is removed into the fuselage structure.



FOR PASSENGERS OR FREIGHT

Interior of a Bellanca Model 28-70 is shown in the photograph. It is a four-seater plane. Right-hand side is for passengers and left-hand side is for freight. The plane has a full and four-seater carrying with metal culture door with complete and upholstery material.



A sketch of the hub mechanism. Wheel and hub are attached to shaft by a key. The wheel is attached to the shaft by a key. The wheel is attached to the shaft by a key. The wheel is attached to the shaft by a key.



The gear fully automatic to retract. Pitch. Pitcher's attitude hub is designed to retract on the standard propeller shaft. The hub is attached to the shaft by a key. The hub is attached to the shaft by a key.

and is an automatic retracting of a tail hub on a wing with a conventional 10 ft. 6 in. wheel, increases the structural control and stability during level flight. The wheel is a 12" x 12" x 12" oleo-pneumatic strut. The oleo-pneumatic strut is a 12" x 12" x 12" oleo-pneumatic strut. The oleo-pneumatic strut is a 12" x 12" x 12" oleo-pneumatic strut.



are fully equipped with ball bearings. The ball bearings are mounted to the shaft of the Bellanca. The ball bearings are mounted to the shaft of the Bellanca. The ball bearings are mounted to the shaft of the Bellanca.

Eligible Propeller Hub

THE LITTLE low line of power plant accessories, the Elgin-Armstrong Corporation of East Orange, N. J., has recently added a fully automatic retractable pitch propeller hub. The design of the mechanism is extremely simple. The hub and blades are allowed to slide forward against a set of springs in response to the direct reaction. A linkage connected to an overhead ball joint carries a torque on the blade end, causing the blade to rotate, a movement toward decreasing the pitch angle. The blade pitch is fixed by the spring and the condition of maximum thrust which occurs at take-off. As the plane climbs and accelerates to top speed in level flight, the linkage gradually increases to the normal level-flight setting. Thus as the throttle is pulled back in slowing down, the linkage reduction in the fixed amount the hub mechanism to set the blades at a still greater pitch in order to decrease specific fuel consumption.

The hub and blades are mounted on the plane in the same manner to any fixed pitch propeller, standard fixed and locking pins being utilized to secure the retracting mechanism. Since the hub is entirely telescopic there are no external controls to be connected.

Fully automatic retractable propeller pitch should have marked advantages over pilot control in combat work where time and pilot attention is at a premium. It should also be attractive to private flyers, who desire the improved performance to be derived from pitch controllability, but who hesitate to accept the responsibility of constant selection.

The first commercially available model of the Elgin hub is designed for horsepower up to 500 and comes in any SAE standard No. 20 propeller shaft. It has been assigned type certificate No. 456.



The Bellanca Model 28-70, designed by Elgin-Armstrong, has a top speed of 110 m.p.h., a top rate of 110 m.p.h. with a top rate of 110 m.p.h. It is powered with a 200-hp. Wright engine of 100 hp.

THE MAINTENANCE NOTEBOOK

In Cooperation with the Maintenance Committee of the Aeronautical Chamber of Commerce

Welding Wagon

FROM Clarence Delina, representative of maintenance at National Airways, Inc., based at the Boston Municipal Airport, comes a photograph of an interesting portable welding outfit which has been found very useful around his shop. The equipment consists of a light tank, electric generator and regulator, oxygen tanks mounted on a dolly which has been made up from an available table and a pair of bicycle Model T wheels. Regulators, torches and rubber hose are hung on convenient parts of the frame proper. This welding outfit is always ready for service at any point in the shop or on the field.



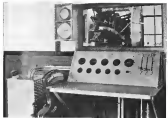
A welding outfit that can be moved right to the job.



A jig for back-boring a crankshaft always at the feet.

Cowl Jig

SHOPS doing extensive metal work on engine cowls may find a useful idea in the jig which has been developed in the shop of Northeast Airlines at St. Paul. The arrangement is extremely simple and can be made up in any carpenter shop from a few pieces of flat board, two or three bolts and some heavy door hinges. As can be seen from an accompanying photograph, the jig may be folded up and stored in very small space. The flat top table on which it is supported is the essential part of the outfit. The cowl can be laid in a convenient position on the table top for laying out the position of fittings, the construction of cover, base, etc.



Front view of the Rembrandt's engine test house at Brookville.

Engine Test House

THE Brookville Test of the American Airways has just put into operation a new engine test installation which incorporates a number of interesting details. The building is a steel-framed building divided into two main rooms—one housing the engine stand and the other the controls. In the test room the engine mount is located on a block of cement, cushioned in 2 in. of cork to reduce the building vibration of vibration. Engine torque is taken on rubber machinery which serves to reduce transmitted vibration. The stand is designed to take engines up to 1,000 hp. Uniform airflow through the engine is assured from a double stack ventilator. Air is drawn downward and upward through one stack and expelled through the other. The stacks are designed so that the flow is independent of engine test direction.

In the control room the ducting, ac-

counting and measuring instruments are grouped in front of a soundproof observation window. The oil tank on the left of the operator and gasoline storage is located outside the building in a underground tank. An electric pump brings the fuel in a gravity tank to the delivery of the observation room from which it is piped through a measuring tank to the carburetor. There is space

available in the observation room for the storage of engine oil and fuel and for the calibrated test properties.

A Group of Lights

THERE are almost no winged hordes of portable work lights in use around airline shops as there are shops

to see them. A group of three selected at random from widely distributed shops illustrates several practical ideas. A tank developed and used at the Boston Municipal Airport is the shop of National Airways in Boston at (A). Here a telescoping steel tube upright carries an adjustable bracket on top, which in turn carries the lamp and reflector. The upright is mounted on a solid cast iron base plate on three leveling casters. It can be pulled nearly around the larger floor, at the same time the heavy base makes it stable enough to eliminate any danger of tipping over. There are four plug-in receptacles near the base from which power may be obtained for electric drills, other portable lamps, etc. Two short pieces of iron welded to the upright make convenient brackets on which the various cords may be wound when the unit is not in use.

A unit of the same type but of a lighter pattern shown at (B) is in use in the crankshaft house of Eastern Airlines at Athens. Here the light reflector is attached directly to the telescoping member of the stand without adjustable brackets by means of a standard connector fitting. Stability is attained by a widespread welded steel tube tripod base. This unit is designed for plugging into portable power distribution boards, or wall receptacles.

An interesting variation (C) comes from the service hangar of American Airlines at Chicago. The low lighting outfit are adjustable brightness and are mounted underneath in smocking brackets. Each lamp is individually controlled by a small switchboard. Power



Three variations in one theme—how to light a shop. (A) National Airways, Boston; (B) Eastern Air Line, Athens; (C) in front of American Airlines' Chicago hangar.

is derived from a regular 110-volt storage battery connected in the base. Thus the unit may be used anywhere inside the shop or out on the field without connection to lighting circuits.

Cylinder Painting Rig

THE Atlanta shop of Eastern Airlines has developed a convenient method of repainting engine cylinder assemblies during overhaul. After cleaning, the cylinders are placed on a rack which turns the top of a wooden table mounted on casters. Supports for the cylinders are provided. The rack holds all cylinders in line, making up in the painting stand which consists of a small tank into which the cylinders may be immersed one by one and sprayed with the new finish. After painting, the cylinders go back onto the portable rack for drying and for return to the assembly line.



A portable rack and turntable unit that is built for cylinder painting jobs.

Emergency Tool Kit

PAN-AMERICAN AIRWAYS' shops go to get in the same sense as the ocean liners. They operate for many hours out of reach with maintenance bases and crews must be prepared to make minor emergency repairs away from home. Each shop, therefore, is provided with a complete repair kit which contains not only a full complement of tools, but also a certain number of spare parts, including one complete engine cylinder, a spare fuel pump assembly, spark control cables and rubber hose, spare control sheets and cans of dope and waterproof paint for emergency patches are included. Laid out clearly in the accompanying photograph one can see the collection of small parts, spark plug covers, gaskets, valve springs, pins, bolts, bracket types, etc., which are carried in a removable tray in the top of the kit. Individual compartments are provided for each item.



Emergency repair kit carried aboard P.A. ships.

THE BUYERS' LOG BOOK

AVIATION's Card Index of New Equipment

This department is equipped to help readers locate manufacturers of new pumps, accessories or materials

AIRPLANE ACCESSORIES

Thip control units

Robey, B. Campbell & Company,
1637-J East 12th St., Los Angeles, Cal.

THIRALLIC equipment for opening flap, retracting landing gear, etc., consists of combinations steel pump and control valve (for cockpit mounting), oil reservoir and actuating cylinder. Complete unit (without pump) weighs 11 1/2 lbs. Pump and valve unit alone weighs 7 1/2 lbs. Pump has pump displacement of 0.40 cu. in. for pressures up to 2,000 lb. per sq. in.

Aviation, November, 1933

AIRPORT EQUIPMENT

Portable units

The Baker Building Company,
Cleveland, Ohio

THE new model CYL portable electric crane is adaptable to airplane overhaul pits involving the landing and installation of engines, etc. Provided in several telescoping beam lengths to cover rails from 8 1/2 to 19 ft. Can be furnished also with towing loop for tractor use. Stowing on all four wheels. All controls concentrated at operating end.

Aviation, November, 1934

ELECTRICAL EQUIPMENT

Battery charger

General Electric Company,
Schenectady, N. Y.

TWO new G-E Tanager battery charger units, one with a three-battery rack and the other for six batteries, have been produced for small shops and intermediate stations use. Units mounted on welded steel frames, have shelves fitted with flexible anti-penetration rubber trays. Cast of stainless steel, including ammeter, fuses incorporated. Occupies only 2-sq. ft. of floor space.

Aviation, November, 1934

ELECTRICAL EQUIPMENT

Battery separator

Electric Storage Battery Company,
Philadelphia, Pa.

A NEW separator material—Edele Meyer—is announced for aircraft storage batteries. It is a form of vulcanized rubber in sheets, perforated with innumerable microscopic pinholes which permit free diffusion of electrolyte but prevent passage of any solid material from cells. Crafted by battery, sulfate or lead. Mechanically strong to withstand vibration.

Aviation, November, 1934

FASTS

Oil seal

National Motor Bearing Company,
1700 Fth Ave., Oakland, Cal.

THE Type BLS and has been developed as an oil retainer for any rotating shaft application. It excludes excess oil and provides positive improvement for oil leaks, or other fluids. Used in supercharger section new turbine engines. Available in sizes from 1/8 in. to 1 1/2 in. shaft diameter. Casts in brass, aluminum-plated, or stainless. Catalog on request.

Aviation, November, 1934

PARTS

Washers

The Wright Washer Manufacturing Company,
Milwaukee, Wis.

ANNOUNCEMENT has been made that a lot of washers especially adapted to the aviation industry has been added to this company's catalog. Washers conforming to both AN-800 and AN-800 specifications available. Can be furnished in any of the materials covered by these specifications—wrought iron or steel, stainless steel, brass, nickel, brass, copper and aluminum.

Aviation, November, 1934

SHOP EQUIPMENT

Value receiving tool

Air Transport Equipment, Inc.,
Newark, N. J.

A NEW value also designed for servicing Wasp and Hornet engines can be used successfully from the front of the engine without necessity of disturbing the cowling. Working parts have been hardened to eliminate wear, and are designed so that the tool will not slip when in use. Fits easily into tool boxes, and is light enough to be used of regularly as plane.

Aviation, November, 1934

SHOP EQUIPMENT

Piston ring gauge

HWS Manufacturing Company,
Philadelphia, Pa.

"RING-MIKE" is a device (developed by the Clegg Gauge Company of Los Angeles) for accurate measurement of G.D. of piston rings. Measures all rings in same center line by means of an accurately calibrated non-removable steel tape. Readings superior on thin center powerful magnifying glass. Made in three sizes for rings 1 1/2 to 4 in., 2 to 5 in. and 5 to 9 1/2 in.

Aviation, November, 1934

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AIRPLANE WHEELS •
BRAKES • PILOT SEATS
AND PNEUDRAULIC
SHOCK STRUTS

THE REASON OF AVIATION SAFETY

AIRPLANE PRODUCTS

BRAKE WHEELS

High and Low Pressure

"Streamline"

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TAIL WHEELS

Low Pressure "Streamline"

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AXLES

For All Wheels

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BRAKES

Mechanically and Hydraulically
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TAIL WHEEL KNUCKLES

For "Streamline" Wheels
Storable and Swivelable
with Shimmy Damper

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PNEUDRAULIC SHOCK STRUTS

Designed and Tested to Meet
Individual Requirements

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PILOT SEATS

Standard Army and Navy Type

BENDIX PRODUCTS CORPORATION
AIRPLANE WHEEL AND BRAKE DIVISION • SOUTH BEND, INDIANA
(Subsidiary of Bendix Aviation Corporation)



"Head into the wind"

SPECIAL MESSAGE TO DEALERS!
Texaco Aviation Products are available to sell. The demand for them is considerable. Write to The Texas Company for details on this important money-making proposition. Address: Department 201.

your engine will always roar a welcome to any wind—when she flies with Texaco!

The exceptional lubricating qualities of Texaco Airplane Oils will reduce frictional drag and increase engine pull. They will maintain pressure under all extremes of operating conditions. Their purity and resistance to sludging will lessen overhaul expense.

Important leaders of commercial aviation such as "TWA," "EASTERN" and "NORTHWESTERN AIR LINES" have proved that MARFAC is the finest lubricant tested for rock-aways. It lubricates effectively. It sustains wear and will not throw out.

TEXACO Aviation Products are available at airports everywhere. And Texaco Engineers will gladly cooperate with you in selecting fuels and lubricants best suited to what you ask your ship to do.

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TEXACO ASPHALT PRODUCTS (for runways, hangar floors, aprons and dust laying)

AIRWHEELS*?

Sure! More and more of the new transport planes are using this safety equipment



The new Lockheed Constellation, built by the Lockheed Development Corporation, has shockproof tires and shockproof, as well as shockproof hydraulic disc brakes.

Point number 1 in "the outstanding features of this new plane"—according to the technical description, is this: "An exceptional high top speed and cruising speed and its low landing speed and ability to land on small fields."

And that's where *Airwheel* equipment comes in. With inextinguishable gears, the size of the wheel or tire makes no difference in flight—but when it comes to quick landings—under all kinds of landing conditions—there's nothing to

match the big, soft *Airwheel** or the reserve power, the smoothness, the sure action, sure release of *Airwheel* disc brakes with non-emerging hydraulic control.

You'll find this safe-landing, quick-stopping equipment on more and more of the modern ships.

When may we tell you all the advantages? For full information write to Aeronautics Department, Goodyear, Akron, Ohio, or Los Angeles, California.

GOOD YEAR

AIRWHEEL IS A REGISTERED TRADEMARK OF THE GOODYEAR TIRE & RUBBER COMPANY

AVIATION: In Goodyear's trade-mark, registered in the U. S. A. and throughout the world, and is used to denote that Goodyear is the exclusive maker of service tires.

When you buy a new ship, specify the Goodyear Airwheel and the new Goodyear Hydraulic Airwheel Brakes.

GRUMMAN AIRCRAFT ENGINEERING CORPORATION

FARMINGDALE, LONG ISLAND

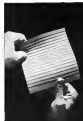


*Designers and manufacturers of Airplanes for the
U. S. Navy*



Exide Batteries are immediately ready to start their stored power because they are completely sealed, thus preventing air from entering the electrolyte. They pass the acid and alkali particles from passing through. This feature combined with numerous other characteristics, such as Exide's famous "Automatic" charging system, makes Exide Batteries the most reliable and longest lasting power source.

Exide



ANNOUNCES A STARTLING IMPROVEMENT IN STORAGE BATTERIES

... Exide Mipor Separators

Aircraft operators know that there is no longer-lasting battery than an Exide—that Exides have always been economical because they hold maintenance costs at a minimum—that Exide dependability is outstanding.

Now, adding enormously to the value of Exide Batteries, is a separator of radically new material that lengthens battery life, increases dependability and greatly reduces maintenance requirements.

The new separators are made of Exide Mipor, a form of vulcanized rubber in sheets. This

material is permeated by microscopic pores so numerous as to permit free diffusion of the electrolyte, and so minute in size as to form a positive barrier to the smallest particles of active material. It is unaffected by the electrolyte and immune to the heat encountered in a battery. It is flexible, resilient, rugged—in fact Exide Mipor is the *permanent storage battery plate insulator*.

These separators have been exhaustively tested for years, both in the laboratory and in service. Cut your maintenance costs with Exides—a good battery made still better.



THE ELECTRIC STORAGE BATTERY CO., Philadelphia
The World's Largest Manufacturer of Storage Batteries for Every Purpose
Sole Dealers of Canada, Limited, Toronto

Exide

**AIRCRAFT
BATTERIES**

WITH EXIDE MIPOR SEPARATORS

*Exide Reg. U. S. Pat. Off.



Jimmy Jones, senior pilot and Superintendent of Operations for General Air Lines, which uses B.G. Men Aviation Kachas Starboard Spach Pumps like those shown above and at the right.

Present in the United States and other countries.

GENERAL AIR LINES

(Western Air Division)

This Line, successor to Western Air Express, is the pioneer mail and passenger airline in the West. It has flown more than 110,000 passengers over 15,000,000 miles without a single fatality or a serious injury. Its route is from San Diego to Los Angeles through Las Vegas (Hoover Dam) to Salt Lake City.



Jimmy Jones, piloting General Air Lines plane and talking to ground station through microphone.

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Contractors to the United States Army and Navy and Aircraft Engine Builders

136 WEST 32nd STREET, NEW YORK

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No. 18 GOES TRAVELING



Another Douglas Transport—production model No. 18—has added a new chapter to the record of these outstanding commercial airplanes. Identified to each of the 41 Douglas Transports that has been produced to date, No. 18 is flying from London to Melbourne with passengers and mail for K.L.M. over each of the airline's regular routes, has demonstrated to



Europe what America already acknowledges as the supreme achievement in commercial transportation.

DOUGLAS  TRANSPORT

PIONEER ROTATABLE AIR SPEED INDICATOR and ELECTRICALLY HEATED PITOT STATIC TUBE



ROTATABLE AIR SPEED INDICATOR Type 735 is an essential instrument, carefully designed to be quickly and easily read in conjunction with other instruments of the flight group. It is usually placed immediately to the left of the Turn Indicator...the Climb Indicator being placed on the opposite side.

The mechanism may be rotated so that when a predetermined air speed has been attained, the pointer assumes a horizontal position on the right hand side of the dial...The Rotatable Airspeed Indicator employs the same high grade mechanism used in all standard Pioneer instruments. Available ranges, 200 to 300 MPH.



ELECTRICALLY HEATED PITOT STATIC TUBE, Type 857D...

positively prevents ice formation at low temperatures, thus insuring operation of the Air Speed Indicator at all times...It is an established fact that within a certain low temperature range, ice will rapidly form on various parts of an airplane. The Pitot Static Tube, because of its small dimensions, may be rendered completely ineffective often before the airplane itself has been seriously affected by the ice load. Concurrently with this ice forming condition, low or zero visibility is usually encountered, making it necessary for the Pilot to depend entirely upon his instruments. The perfection of the Heated Pitot Static Tube makes possible reliable airspeed indication, regardless of temperature. To provide for various methods of mounting, Pitot Static Tubes, type 857D are offered in three models. All three models have the same Pitot Static section as illustrated above.

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• The Consolidated P-30 is the highest performing Two-Seat Military Pursuit Airplane of its type available in the world. It embodies advanced features in design, construction and utility, with unparalleled maneuverability and high speed at great altitudes.

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• Both the P-30 and the A-11—the result of three years' intensive engineering development—have been tested extensively as service equipment. Their use, in sufficient numbers, will place our air force in a definitely superior position to that of any other nation, and will add tremendously to our Air Corps' striking power.

THE CONSOLIDATED AIRCRAFT CORPORATION, BUFFALO, N. Y.

STRAIGHT FACTS FOR STRAIGHT THINKING

By *munster*

NEWTON D. BAKER, CHAIRMAN
NATIONAL CITIZENS COMMITTEE

1. It is true that billions are being spent by the Government in order that people may not die of cold and hunger.
2. But these billions, divided among the families in need, average for each family only about \$24 a month.
3. And 70% of the free hospital services in the United States for the needy sick are provided by voluntarily supported hospitals. The sick among the unemployed number 40% more than among the employed.
4. Likewise public health nurses, also supported by your voluntary gifts, report that 66% of all their visits in 1933 were in homes unable to pay for the service rendered.
5. 30% more children have had to be removed from their own homes and cared for by voluntarily supported children's agencies.
6. Two-thirds of all the arrests for crime involve persons between the ages of 15 and 24 years. Millions of boys and girls living under conditions destructive to character need the character-building services of your recreation agencies.
7. A man may die of despair, as well as of hunger, for suicides, numbering 15,366 in 1928, grew to 20,927 in 1932. This shows that more and more people are coming to value the only kind of life they are able to attain.
8. America cannot be rebuilt by relief measures alone.
9. Your local community chest needs your support during this year of rebuilding human hope and morale. It supports hospitals, clinics, child-care organizations, character-building agencies and many other social services.
10. When you give in your city, you strengthen the forces of civilization in the neighborhood in which you live.

1934 MOBILIZATION FOR HUMAN NEEDS

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AND LAND 75 FEET FROM WALL STREET



At the Wall Street Pier (below), on the East River, New York City, large transport planes are handled on the floating rubber-tide stage. Small seaplanes are the 400-foot floating airport of the jet. Another floatable stage was built at 1934, Street and Riverside there last year (below). A third New York City stage will be installed at 200 Street and East River.

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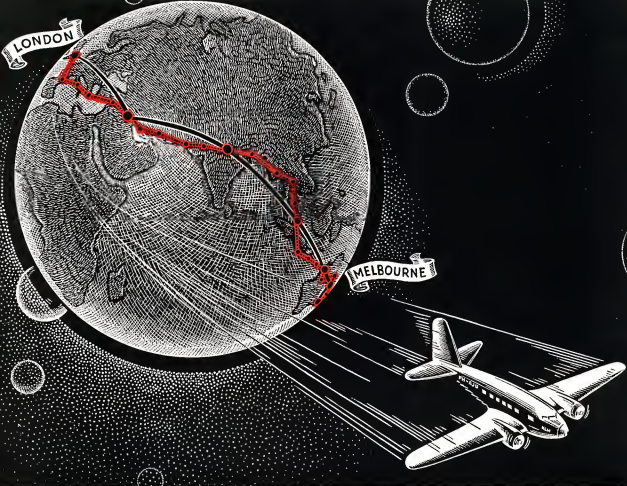


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